

September

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Story of
THE WEEK AT
Cleveland

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Overseas Division

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D. J. BARRETT, Jr., of Worcester Airport, believes in training students in modern ships. "The kind they will want to buy and you will want to sell," he says. And so, this summer, he switched to Fairchild KR 34s. In the first month's operation his three regular Fairchild KR training ships had 107 hours, 68 hours and 55 hours respectively in the air. His total instruction ship has scarcely been needed.

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If you fly a Curtiss ship or have a Curtiss engine in your place... you are always within easy flying distance of complete spares, parts, and necessary service.

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ABOUT LEFT

One of the 47 refueling contacts in the dramatic flight. The mastery of the two Challenger-Robins was a big factor in the success of this dangerous operation.

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Curtiss products — test! Forest O'Brien (top) and Gale Jackson (bottom)... members of the St. Louis Robin.



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Daily, under all conditions, this Jackson went out on his precious "outcall" to inspect and tune up the Challenger engine... the pivotal member of the new famous quartet of Jackson, O'Brien, Robin and Chabinger. Dramatic close-up shot made high in air on fifteenth day of flight.



AT LEFT

"Grounding" the Challenger engine just before the start of its battery making flight.



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THE OLDEST AMERICAN AERONAUTICAL MAGAZINE

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EDWARD P. WARNER, Editor

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Civilian Stunting

IT IS TOO EARLY to make any full and final appraisal of the Cleveland Races and their results. It is not too soon to give an enthusiastic approval of the general effect. The management deserves great praise, both for the program as a whole and for the arrangement for handling the crowd, but there was one event, almost daily repeated, upon which we can only bestow unreserved condemnation.

Each afternoon stunting teams of civilian pilots in aircraft of commercial types were forth to display their prowess. The piloting was splendid. The machines well proved their structural strength and their maneuverability. The crowd was thrilled almost as much as by the military formation flying. The spectacle was one which had absolutely no place at the National Air Races, and we hope that it will never again be seen there.

As a demonstration of the capacities of an airplane for the benefit of a prospective purchaser who is already a pilot and is well acquainted with the general behavior of aircraft in flight, aerobatic flying has much to recommend it. It is a proper part of any such demonstration of a light sport machine designed to stand rough handling. As a demonstration of the methods of aviation and the general capabilities of aircraft before a crowd 99 per cent of whom are not pilots and a majority of whom probably never have even flown as passengers, it rates below absolute zero. When a string of outside boys in caps, those who fly themselves may understand and admire, but each succeeding inverted flip-flop adds to the number of lay spectators who sincerely resolve that they will take all their aviation hereafter at second hand and at a safe distance. "If that sort of thing is a part of flying," is the comment, "aviation tells me that it cannot be much fun."

One afternoon, following aerial exhibitions which included the extreme in aerobatics, the race announcer asked that all who had been up in a plane raise their hands. Perhaps one-quarter of those assembled indicated that they had taken flight. The announcer then asked that all planning to take rides after witnessing the exhibi-

tion's exhibition raise hands. The additions were scanty.

It is difficult to blame either the pilots or the mesmerizing companies taking part in stunting displays. Competent operators there as elsewhere, and if one commercial machine is to survive the crowd by level evolutions, road crosslanders are usually eager to prove that their products can do the same. Oddly by co-operative action can the evil be met. We urge it upon the commercial manufacturers' section of the Aeronautical Chamber of Commerce, with all the force at our command, that they jointly agree that stunting exhibitions on these public occasions are opposed to the interest of the airplane industry and that they pledge all their members to refrain from participation. We urge that the Department of Commerce use its good offices to the same end, and that the National Aeronautic Association do its best to discourage anything in the nature of stunting contests at sanctioned meets.

The idea of using aircraft as vehicles has to be sold to a large and stupid and infinitely somewhat timid public that has not yet been tamed. Man alone in discourage aerial aerobatics is one of the obviously necessary steps to that end.

//

The Ladies Come Through

WHEN THE PLAN for a women's air derby was announced, there were many doubters. Metropolitan newspapers passed with individuals both in the aeronautic industry and outside in pessimistic forecasts of disaster and of a disastrous public reaction.

Well, the derby has been run, and it was worth while. Despite the tragedy which shadowed its early stages, disheartening to fatal accidents always are, the event came as a success. Where six months ago there were but three or four women in America who held the transport license, it was possible last month to assemble in San Diego a score of feminine pilots who could satisfy the exact committee that they were qualified for the event,

and two-thirds of them come through to the finish over a course of more than two thousand miles. That criterion of the criteria were very limited in safe consistency experience or in any kind of experience with readiness of the type that they flew in the race. Frankly candid some of us great athletes in the first days of the competition. That they came through to the finish successfully in spite of these limitations stands, now that the event is over, as a testimony to their mental attitude for flying.

It would have been easy for the women's derby to become a sort of an annex to Hollywood. That it did not, and that unglamorous spectators were few, is a tribute to the good sense and seriousness both of the managers and of the contestants. It will still be easy to err in the future. If there are to be repetitions of the women's events, both cross-country and closed circuit, which have been held in connection with Cleveland's gala week, we will take it for granted that there will be they must be considered with due measures to the love and the path. It is very much in the background. The safety of the contestants demands that, and in does any sound promotion of women as a sport. Rules should be stiffened up, rather than relaxed. Over-confident women must be protected against themselves. This year's trial was (freely as experiment. The women who started at Santa Monica on August 18 and finished at Cleveland on August 26 taking three as a whole, set a good example for their successors.

//

The Saratoga Goes on the Night Shift

IT IS OUR REGULAR POLICY to abide to military and civil happenings only so far as they directly affect the aviation industry. We undertake no general treatment of military policy, and no political arguments. In the recent accomplishment of the aircraft carrier *Saratoga* in making repeated landings abroad at night by the aid of a few lights placed on the deck of the ship, however, there is nothing political. Even though the accomplishment is an essentially military one, and even though a proper military agency on aircraft carrier operations governs discussion of the details of the method, we cannot without our enthusiastic admiration. It is difficult to realize without having been on board an aircraft carrier the extreme deficiency of technique required to land on board a moving ship, rolling and pitching as it travels through the sea. The action of the pilot and the action of those on the deck must be so harmonized that three minds work as one. The difficulty of the problem is infinitely greater when the ship and its surroundings are not visible as a whole through the approach.

Although there is no apparent immediate commercial use for the experience so gained, the provision of landing

decks on a commercial liner has at least once been seriously considered in preparing plans, and is likely to become a live issue again. In any case, however, and whether or not commercial airplane carriers are built, the landings made on the *Saratoga* at night serve as a fresh reminder to the general public that night flying has become a commonplace thing. Even after five years of successful night air mail operation, a non-piloting passenger may still regard with some terror the idea of having to get down in a comparatively strange and dimly lighted field after dark. The knowledge that such landings can be made in darkness on so suitable a surface as the wearing deck of a ship should do much to encourage him. The officers of the *Saratoga*, as well as the high command of the Battle Fleet Aircraft Squadron and of Naval aviation in general, are entitled not only to the warm congratulations but to the thanks of civilian aeronautical interests for this demonstration.

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A Wise Precaution

PARACHUTES have been required of all participants in the air derby, and are strongly urged for participants in all the other National Air Race events as well. It is an impression that this is the first time that such action has been taken by the National Air Race sponsors.

They have assumed a most commendable stand, and we say this notwithstanding the fact that one attempt appears to have been made to use a parachute from too low an altitude to permit of its functioning. Modern parachutes have been developed to unquestioned reliability and their ordinary value is heightened under conditions of race or low start emergency, such as are bound to exist during the military air. Adherence to the Air Race Committee's request may not actually have any effect in any particular event. There may be no occasion to use parachutes. Still, so reasonably available safeguard should be neglected.

Unfortunately, the National Races have been held at the cost of several lives in past years. Unfortunately, too, circumstances attending some of the accidents would have made every safety device, including the parachute, of no avail. However, the normal arguments for the use of parachutes gain in impressiveness at a time when additional rules, such as exist in racing, are assumed.

There will exist on the part of some good pilots a prejudice against parachutes under any circumstances. Their attitude is hard to understand. Aerially in general are far safer now than ten years ago, and there are many operations in which the prospect of needing a parachute or being able to use it successfully is so slight as not to justify the added weight and inconvenience, but where the

parachute saves down apparently a real risk, we are strongly in favor of it, whether in commercial or sporting pursuits. Consequently racing and such closed-circuit racing as well, falls in that category. It is to be hoped that the recommendation of the National Air Race committee was generally followed and that it will establish a standard practice for the future.

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Free-for-All Racing

FOR THE FIRST TIME in several years the free-for-all at Cleveland was true to its name. The old rule against paying military and civil pilots against each other, which has stood for ten years or more, has been laid aside. We hope it will never return.

It is not the policy of the War and Navy Departments so much as it is general conditions that have changed. Two years ago the military had neither the resources nor the inclination to build for racing.

A civilian owner at Cleveland furnished the best of evidence that the commercial builder can enter against unscrupulous competition, and can sometimes win. It gives his engineers an interesting exercise, and opens them to new effort which may later reflect well in the stock market put into production. Is it worth while?

So long as the racing is taken so much seriously than at Cleveland, it is well worth doing. The military machines competing there were straight pursuit types. The non-military entries, though hardly suitable for transport, were not so specialized nor so utterly expensive as to fall of appeal to a small but growing class of wealthy young men who seek extreme performance in a sport machine. Everyone got experience. The prize-winners got valuable advertising for machines at least approximating commercially suitable types.

The danger is that we shall not stop here. "Free-for-all" is a two-edged sword. One Schneider Race is enough for the encouragement of design for sustained speed. As competition grows hotter the temptation is to put leading speeds up and up, to run down cockpit openings, to sacrifice safety, vision, durability, every thing else to the quest for miles per hour. The end is the men pouring out of money upon racing plans having no other possible use, and a withdrawal of the airplane industry from the open course as the automobile industry, taken as a whole, has withdrawn from the track. Before next year there should be enough some knowledge that will insure some practical utility in the planes entered. Limitation on engine displacement alone will not do. There are various alternative possibilities, but we do not attempt at the moment to discriminate among them. We do hope that the matter receive early study, so the end that building for race and entering there may continue to be worth the industry's while.

Nine Starts ... Eight Successes ... No Disasters

THE HEADLINE tells the story. The very simplicity of the statistical statement of the rapid airship's Trans-Atlantic performance is impressive. It reads an understatement. The wonderful flight around the world, just completed, was hardly required to prove the case. The crossing of the Siberian wastes and of the North Pacific coast as a super-demonstration. There is no urgent need to present for confirmation road-the-world airship service, but a credit that can perform such a trip with such ease and apparent ease inspires new confidence in its capacities for consistent travel on regular routes of more modest length.

Towed by airship between America and Europe has become a normal thing. Like traveling by airplane between Los Angeles and San Francisco. The passenger list of the Graf Zeppelin includes the names of business men who are there not because of any leaving look for airships but solely because their affairs require their presence on the other side of the Atlantic with a minimum loss of time. They would be amazed if there were any delay, either in starting or in arrival. The whole world would be astonished at anything worse than delay. Success is taken for granted. In proportion as the airship ceases to be a vehicle of romance, it is becoming a vehicle of transportation.

The smoothness and uniformity with which the Graf Zeppelin has been shuttling across the Atlantic, reinforced by a circumnavigation of the globe in 21 days and seven hours total elapsed time with only twelve days of actual flying time, furnishes a noticeable contrast to the varied record of trans-oceanic operation with heavier-than-air craft. The airship on the other hand would undoubtedly score a deplorable failure on certain short inland routes which the airplane now serves with regularity and efficiency. Never has it been more evident that the friends of both types will be gains of outrageous folly if they trust each other as enemies as a war of extermination. Airplane and airship need each other's helping. They operate in different fields with a substantial measure of overlap to be sure, but with not enough to make them presently competitors. In choosing aircraft for transport we may well adopt as a motto, though with changed application and different purpose, the first part of the Socialist creed: "From each according to its ability." "The forces of nature," said the great Count Zeppelin, from whose undoubted spirit the rapid airship emerged and whose name it bears by the way, the latest of the line, "cannot be eliminated, but they can be balanced against each other." We shall be wise to choose the particular type of balance and the result that does the balancing, with due reference to the nature of the particular forces to be met. Let us have no dogmatic devotion to standardized solutions for our aerial problems.

Technical Aspects OF THE CLEVELAND EXPOSITION

By LESLIE E. NEVILLE

AN ANALYSIS of the exhibits at the National Aeronautic Exposition, held August 24 to September 2 at the Cleveland Public Auditorium, is helpful in setting the recent trends in airplane and engine design. Although the airplanes in the exposition hall were fewer in number than those at some of the previous shows, the group was representative of the present status of the American aeronautical industry.

The Cleveland exhibition is exceptionally well adapted to any type of exhibit, particularly one of airplanes and both city and show management are to be commended on their forethought and consideration in making preparations for the display. Unfortunately the attendance was somewhat below standard but this was more than counterbalanced by the record showing at the airport. There is probably some question as to the advisability of conducting an aircraft show at the same time as the air races and at so great a distance from these activities.

As to the presentation of the exhibits most of them were attractively arranged and well spaced and balanced. Decorations were simple but pleasing. Several companies showed unpowered fuselage structures and one mid-engine was displayed. It is regretted, however, that there were not more exhibits of this type and it is hoped that American manufacturers will realize the value of such displays as shown by the European expositions.

One of the early Wright biplanes and the two-cylinder Laeuer engine, from which the Wright Whirlwind was evolved, lent an historic aspect to the exposition.

Probably the dominating theme of the exhibit was the response of designers to the demands for increased performance in all classes of airplanes and most of the developments seen could be attributed directly or in-

directly to this. This design trend was reflected also in the engine and accessory exhibits.

Increased performance depends in general upon two factors, greater power and cleaner design lines in aerodynamic standpoint, and most of the airplanes exhibited showed evidence of development in one or both of these. Increasing power is limited of course by the engines available in a given class while that of increasing aerodynamic efficiency is to a great extent a problem of decreasing drag.

ENGINE MANUFACTURERS are trying to meet the demand of the airplane manufacturers in this respect and the plane manufacturers are adapting their designs to the engines of greater power as rapidly as is practicable. The engine designers are striving not only to produce power plants of greater output but are also attempting to combine with the plane manufacturers' efforts by designing engines of lower frontal area and therefore less resistance. It is interesting to note that this was the first aircraft show at which no airplanes powered with the Curtiss CX-5 engine exhibited.

Among the design features contributing to aerodynamic efficiency were cowlings of the NACA type, streamlined fairings of several distinct types for landing wheels, fillets to streamline wing roots and engine nacelles into fuselage, fairings of circular or elliptical section, movable and retractable landing gears and landing lights. Tapered wings were not so much in evidence as at previous shows probably because of the difficulties involved in their manufacture.

A number of the aircraft at the Cleveland show were equipped with cowlings of the NACA type but no streamlined interior design seems to have become

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accepted. The practice of building these cowlings is taken so far as to permit of removal without disturbing the propeller in quite general but none of these installations were provided with any means of adjusting the size of the window opening at the rear.

Streamlined landing wheel fairings have become quite general judging by the number of airplanes equipped with them at the exhibition and in the field. The pre-



Above: A detail of the interior of the Curtiss CX-5 radial monoplane. Left: The six-cylinder "Whirlwind" six-cylinder in the Cleveland air model section rated at 320 hp.

ailing type was a tapered thread covering as much of the wheel as practicable, constructed of steel tubing or strip and covered with fabric. One exception found on a Monocoupe plane, was a fairing of modified oval shape fastened by metal strips just behind the wheel. The average weight of these single units is approximately 4 lb.

Many airplanes thus at any previous show were fitted with fillets to fair the attachment of wing and fuselage and, in one case, these fillets were used to provide a portion of the skin for the fuel tanks, which also extended into both wing roots.

The difference of opinion of designers as to the desirability of increasing efficiency by the use of various types of fuselage sections is still in evidence. Where some of the new fuselage designs at the Detroit show indicated an attempt to derive a certain amount of lift from this unit, those at the Cleveland show seemed to suggest an effort to reduce drag by the use of elliptical cross-section sections.

While it is not new, the cambered landing gear of the General Anson and Cadet sportbuses seems worthy of mention and it is also interesting to note that the retractable landing gear of the Eaglehawk "Buller" has been somewhat redesigned. The principal modification is in the direction of travel of the wheel barrens which in new form are all.

A large number of low drag landing lights, both of



the built-in-the-wing type and the retractable type were shown, some installed on airplanes and others among the accessory exhibits.

An invention and an eight-cylinder engine type of engine, both intended to facilitate streamlining and increase versatility in airplane designs were shown and will be discussed in detail later.

CONSTRUCTORS the airplanes at the show from the structural standpoint, the progress of metal construction is still noticeable as it was in Detroit last April and welded steel tubing remains conventional practice in fuselage construction while the majority of wing structures are still made of wood. Fabric remains the favored structural material but a large proportion of the new designs at Cleveland employed plywood covering. There were no wing structures of welded steel tubing and case of steel and aluminum alloy construction but there were several of wood and aluminum alloy.

There seems to be an inclination on the part of designers to separate the pilot from the passengers in closed airplanes intended for a single pilot. This idea has its advantages but it would be a decided disadvantage if for any reason the pilot was incapacitated and no one could take the controls is time to abort disaster. One design at each of the last two shows had engines built into the wings. Some progress is being made toward solution of the visibility problem, but there is still much to be done in this direction.

With the increasing number of light training planes several problems seem to have arisen. Economy dictates that one set of instruments should be sufficient for pilot



The Pratt & Whitney engine, shown in detail, is a six-cylinder radial engine with retractable landing gear mechanism.



and student and this is solved automatically in planes of this type having side by side seating arrangement. These appear to be in the minority, however, and various devices have been developed to meet this and to permit of closer communication possibilities between student and pilot in the tandem seating type.

One of the first of these was the single cockpit for tandem seating arrangement. This was not in evidence in any of the airplanes at the Cleveland show, and it is interesting to note that one manufacturer whose design



above—Sketch of the first school and student tandem arrangement of the tandem type. Model—The McCulloch as it was exhibited at Cleveland. Below right—Sketch showing the construction of a tandem aircraft and better on the lower portion of the fuselage of the airplane which is present with the tandem design.



was characterized by such an arrangement has discarded it entirely, made two separate cockpits and provided a small glass window for the rear cockpit.

In some cases no instruments are provided for the student as standard equipment. Other training planes have the necessary instruments mounted above the cockpit covering and in an effective installation, the instrument panel was built into a console in the center section. This also allows vision ahead.

Most of the training planes now have detachable seats to avoid the dangers resulting from student "freezing" and a detachable rudder bar was shown as part of the equipment of the Swallow TP.

Tail wheels, brakes and other shock absorbing mechanism are a general consideration on private and some commercial planes, while those of the general purpose types may be equipped with these devices at the option of the purchaser. Although propeller equipment is usually optional the airplanes at the show were almost evenly divided as to wood and metal types. No noticeable tail shafts or wheels were noted at the exposition. Parachute seats also are coming into more general use.

For changes have been made in the general design of landing gear but several designers are using wing struts to obtain greater strength. A new tripod type of

landing gear has been developed by the Waco Aircraft Company and is being used on all of its planes. The control and external appearance and finishes of airplanes seem to be following automobile practices closely. This also applies to interior hardware and instrument equipment.

In point of numbers, monoplanes predominated over biplanes at the Cleveland show. Comparing this exposition with that held in Detroit the proportion of these winged planes at Cleveland was far greater. This was due of course to the addition of the Boeing Model 80-A to the transport class and the Hiss Monoplane in the medium weight tri-winged class. The first two and twin weight tri-winged planes were shown indicate that designers are still interested in this type. Many of the new designs are low and center wing types, and a number of planes of this type are offered for training purposes. The demand for a small airplane for private



use is also being met by designers according to exhibition at Cleveland. A comparison of the Detroit and Cleveland shows regarding numbers of the various types exhibited is shown in Table 2.

An analysis of the airplanes exhibited on a basis of gross weight also is interesting. Beginning with the weight range under 1,000 lb., two planes were represented, both being of the single wheel light type and both monoplanes. It is interesting to note that the 1,000-2,000 lb. range had the greatest representation in point of numbers. Seventeen of these (all of the two plane

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type) were exhibited and the majority of them were intended for training purposes. Sixteen airplanes in the 2,000-3,000 lb. range were displayed and this is the first exposition at which there were fewer of these planes than those of the 1,000-2,000 lb. class. There was a smaller proportion of airplanes of the medium than those of the general purpose type in the 3,000-4,000 lb. class, indicating that this type has been supplanted to some extent by the two plane type of similar design and also that the needs of airplane purchasers have become more clearly defined.

Both of the aeroplanes and both of the medium weight tri-winged planes fall into the fourth range of 3,000 to 4,500 lb., of which there were eleven. The last planes in this class are largely of the sea plane type. Only one of these planes was a biplane.

The fifth class ranging from 4,500-6,500 lb. was represented by six planes of the single engine land and transport type. Four of the six were monoplanes, one a biplane and one a sesquiplane.

The three tri-winged transport planes exhibited fall into the range of more than 6,500 lb. and had passenger capacities ranging from fourteen to eighteen. Two were monoplanes and one a biplane.

OF THE FORTY-FIVE AIRPLANES at the exposition and the fourteen trainers at the airport at least four were rushed to completion by their manufacturers for the exposition. Two of these had not been completed in time for test flights while the remaining two had been flown.

One of the most interesting and original designs at the exposition was the Model 207 Mail Transport Plane exhibited by the General Aircraft Corp. The general specifications for this plane were submitted after a questionnaire had been sent by the General engineering staff to a large number of mail operators in an effort to obtain a consensus of opinion as to the requirements for this type of service.

The model 102 is a sesquiplane having a rigid fuselage, alloy monocoque landing gear, semi-cylindrical monocoque section and powered with a Pratt & Whitney General Hornet engine. It was the only airplane at the show powered with a geared engine. This airplane is designed for heavy load carrying, having a payload of 2,000 lb. with a useful capacity of 3,150 lb. Its maximum gross weight is 6,230 lb. The wing span is 35 ft. and the overall length, 35 ft.

Spruce beams and aluminum alloy ribs with fabric covering are employed in the wing construction with



View showing the interior construction of the Mailplane



The struts showing the new type of landing gear had filed on planes of the Waco Aircraft Corp.

the extension of the center section which has been trapezoidal aluminum alloy beam. An aerial section, having medium high lift qualities is employed in the wings which have tapered and widened out tips.

Unusually complete instrument equipment is provided and sliding rudder pedals with three possible adjustments for stature are furnished. The tail wheel is a non-adjustable novel type having a coil and spring to keep it in normal position and is fixed in travel. Oil shock absorber mechanism for the tail wheel is directly attached so that it does not come with the wheel. The plane is fitted with covering of the N.A.C.A. type having the oil tank built into the covering to provide cooling. An exhaust collector ring is also provided and leads the exhaust through a tube streamlined into the lower portion of the fuselage and acting as both a heater for the interior and a silencer.

Another interesting airplane at the show was the Great Lakes 441 Aeroplone powered with two of the latest model American Curtiss engines. This craft has been developed to meet the demand for a low priced aeroplone of reasonable price which has long existed using power type.

The 441 aeroplone is a four plane biplane type, having engine mounted in streamlined nacelles between the wings which are of equal size and monocoque in plan form. The wing construction is similar to that of the Great Lakes training plane, consisting of wood spars, aluminum alloy ribs and fabric covering. The Gorington 358 aerial section however is solid. Both structure and sheathing of the hull are of aluminum alloy and riveted. The hull is entered through a hatch directly above the rear seat and a portion of the seat back folds down to form a step while entering. A passage way between the nacelles for the mounted landing wheels leads forward to the control compartments which contain two other seats. The shock absorbing mechanism on the landing gear is hydraulic while the resulting mechanism is operated by cables.

The tail gear is well above the water line to avoid damage and is constructed of corrugated aluminum alloy. The craft has a wing span of 29 ft. 3 in. and a gross weight of 3,200 lb. The weight empty is 1,900 lb. Provision has been made to carry water ballast of fuel.

Representing the increasing class of low and center wing monoplanes, the Emsco Circus was another of the airplanes to make first appearances at the Cleveland exposition. This airplane is of the semi-braced, center wing type and is intended for sport and training purposes.

It is powered with the American Curtiss engine, has a wing span of 36 ft., an overall length of 28 ft. 10 in. The weight empty is 1,090 lb. and the gross weight, 1,640 lb.

Conventional practice is followed in the construction of this plane which is the most normal product of the Finner Aircraft Corporation. The entire plane is well streamlined and an angle fairing at the junction of wings and fuselage provides space for two 15 gallon fuel tanks in the wing roots. In a further attempt to reduce parasite drag, streamlined heads are placed over the landing wheels. The engine exhaust, which consists of a metal steel manifold tube, runs to a point behind the rear cockpit, is suggestive of the types used on racing automobiles.

Although not completed in time for the show, the new Travel Air low wing monoplane is noteworthy because of its excellent performance at the airport. Two models of this highly interesting design were shown and flown at the airport by pilots Davis and Clark of the Travel Air Company. These two models were identical in gross design and construction, varying only in power plants and resistant aerodynamic possibilities. One of the planes was powered with the Wright New Whetstone Nine engine in an A.C.A. cowling installation while the other was designed for the new Chevrolet engine, a six-cylinder inverted in-line or cooled type which, like the airplane, made its first public appearance at the Cleveland Airport during the race.

The new airplane is a single seat type, with no sections contributing to high performance omitted. The fuselage is of elliptical transverse section and is constructed of plywood on a welded steel tube structure. The wing structure is of the conventional type and is also covered with plywood. It is internally braced by streamlined struts. The lower portion of the cockpit oc-



This right cylinder appears "Gem" engine exhibited by the Electric Machine Engine Company

cus a continuation of the fuselage lines streamlined to a point. Fairings are provided for the landing wheels.

Another aircraft not new but publicly exhibited for the first time at the Cleveland exhibition was the McCulloch Dauphin, "Vespene". This craft is a four place, twin engine, monoplane amphibian, powered with two New Whetstone Five engines and designed also to meet the demand for a small amphibian. It was exhibited by the Cadillac Aircraft Corporation.

Engines are mounted in housings faired into the upper surfaces of the wing which, in the initial product, is constructed of wood and plywood with the exception of the center section which carries the engine mountings. The single step hull is built of wood with streamlined aluminum alloy sheathing. A more extensive use of metal is planned by the designers, however, in future models. Oleo shock absorbers are used on the landing gear, which is retractable by means of a drum and cable mechanism controlled by a crank in the cockpit. The wing spans is 40 ft. and the overall length 35 ft., while the weight empty is 2,075 lb. and the gross weight 4,100 lb.



Streamline installation on General Cadet Sportboat

ONE CLEARLY DEFINED GROUP of airplanes at the show was that of new models developed from existing designs. This included the Curtiss Thrush, Boeing Model 80-A, the General Aeroplane Cadet Sportboat, the Travel Air 10-B, the Golden Eagle Owl and others.

The Thrush is a six place cabin monoplane bearing a marked resemblance to the Curtiss Robin, and also general with the 120 by Curtiss Challenger engines. The airplane is also available with the Wright New Whetstone Seven engine and it was this model that was shown in the exhibit of Curtiss Flying Service. The Wright powered Thrush is an extremely braced type, of conventional construction and furnished with complete instrument equipment. It has a wing span of 40 ft., an overall length of 32 ft. 6 in., a weight empty of 2,160 lb. and a gross weight of 3,590 lb. A fuel capacity of 110 gal. is provided.

While few fundamental changes have been made in its design and construction, the internal arrangement has been changed considerably in the Boeing Transport Plane Model 80-A. A passenger capacity of 18 persons is now provided. The wing span has been increased to 85 ft., the length to 55 ft., while the weight empty is now 36,413 lb. and the gross weight, 57,900 lb.

In order to extend its line of models into the open

training plane field, the General Aeroplane Corporation has developed the new trainer, which is a modification of the Aerostar. This plane is a two place tandem monoplane of the externally braced type. One of the features of this airplane is a rectangular cut-out in the center section of the wing, placed on the upper surface with a transparent material to provide visibility there. An instrument board is mounted in the space formed by the thickness of the center section in the forward side of this wing. This makes it possible for the instructor to be seen by both student and pilot. A large



Alexander clutch showing reduced landing gear of McCulloch Dauphin. Model-B general amphibian showing installation of the two 100-hp. Ford V8 engines

take type of control makes it possible to operate the switch, mounted on the right of the instrument board, from both cockpits.

Several modifications have been made in the Wright powered Travel Air 10-B cabin monoplane, which is somewhat similar to the 6,000-B model, but with accommodations for four instead of six persons. The shape of the windshield has been changed in the 10-B to provide greater visibility. The 6,000-B is now available with four seats and a coach instead of six seats as formerly.

A number of refinements have been made in the design of the Golden Eagle "Owl". This is a development of the original Golden Eagle monoplane. In the new model, a detachable cowling with an A.C.A. cowling, and streamline wheel fairings have been added.

One of the late arrivals at the show was the re-engineered cabin monoplane, exhibited by the Finner Aircraft Corporation. This plane has accommodations for six persons and is powered with three Chrysler 25-5 engines. It is constructed conventionally and has a wing span of 52 ft., an overall length of 30 ft. 10 in. and a gross weight of 4,280 lb. It was the only airplane at the show having an overhead control stick and it was voted by the designer that overhead controls of the wheel type were to be installed in forthcoming models.

Two airplanes are to be used increasingly by executives as offices is indicated by the fact that several crafts engaged for this service were found at the exposition and at the airport. Among these was the Lockheed Educational 4 seater plane. Wasp powered model of the same general type as the "Vogel" and the "Air Express." This plane has a desk and space for a portable typewriter, as well as the revolving lounge, seats and a telescoping lavatory compartment.

The semi-closed monoplane exhibited for the first

time in Cleveland by the Standard Aircraft Corporation also was provided with a desk and space for other office equipment. An airplane designed especially to accommodate a dispatch was displayed at the airport by the Fairchild Aviation Company.

Another indication of the variation of the designers to adapt airplanes to use as offices was found in the conventional adaptation of the Vought Corsair. A sliding windshield covering a large portion of the cockpit, is closed position, has been made part of the equipment of this plane so that papers can be handled in that cockpit.

Another adaptation of military design, were to sport their commercial use, was found in the exhibit of United Aircraft and Transport, Inc. This was the Boeing Model 100, a high performance fighter now straggled as military equipment and offered in the commercial market.

In addition to the Model 100, the Boeing company showed the Model 40-B4, a relatively new design. This



plane, which is known as a mail transport plane, is designed for two distinct types of service. An open cockpit at the rear is provided for the pilot, while the closed portion forward can be used to accommodate four passengers or an equivalent load of mail or express. A unique design feature of this plane that would be useful in other aircraft, is a ring welded to the tail structure to facilitate lifting the tail or bracing it to a dolly.

The tail fin is in the field of high wing planes were the Island Sport and the entire wing monoplane exhibited by Standard Toolcraft. The Island Sport is an open monoplane of the parasol type with seats in tandem. It is powered with the Le Blond Sixty engine. The Toolcraft plane, which is not yet in production, is a two place, side by side seat type, having several novel features. The fuselage is a Warren truss type of wood structure with plywood covering and the wing is of conventional wood and fabric design. A portion of the fuselage covering is cut away where the wing struts are attached and it is possible by the use of transparent material, in the lower surface of the wing, to see the landing gear from the cockpit. Future plans of this type will be built of metal, according to the designer. The Le Blond Sixty is the standard power plant.

IN ADDITION to the airplanes already mentioned, Kinner powered models of the American Eagle, Walker Tour-a-lap, Arrow Pursuit, Deuster Wake Red, Fleet, and Seafowl TP were shown and the Kinner Air Coach with three Le Blond Sixties and the Autocrat Kline low wing monoplane (Landing Arrow Speed, Drive V-3, Roving 10-B, with Le Blond Sixties. The Curtiss Challenger powered Commandaire was shown by Curtiss Flying Service, Inc. Models powered with the five-cylinder Wright J-6 engine were exhibited by Alexander Aircraft Corporation, Fairchild, American Eagle and

Waco Aircraft Corporation and the new General Aeromarine powered with this engine was shown. Bellanca, Fokker, Ryan and Stearman planes powered with the six-cylinder J-6 engine were displayed and Waco and Curtiss showed engines with the seven-cylinder J-6 engine. Pratt & Whitney engines were installed on the Boeing 80-A, Boeing 100, Fairchild, Fokker, Ford, Hamilton and Wright planes.

The Hiss Warner engine was installed in the Albatross Argo plane and the Lambert engine was shown in the Star Corsair.

Two seaplanes and one flying boat were exhibited, but were not new developments. The American Klean, The Bellanca CH-300 Seaplane and the Eastern Flying boat have been described. Two gliders were displayed by Herschle Altam of Cleveland.

THIRTY-ONE ENGINES were displayed at the Cleveland exhibition and it is significant that only one of the new designs has been described. The water-cooled type. Of the remaining thirty air-cooled models, twenty-four were of the radial type, three were in-line-cylinder-in-line vertical, one was a converted rotary, one



The piston and connecting rod assembly on the six-cylinder engine.

a six-cylinder-in-line inverted and one an eight-cylinder opposed type.

The two last mentioned were new designs that were exhibited for the first time at the show. As previously mentioned, both of these new designs have been used in an effort to facilitate assembling in their installation and to increase reliability ahead of which is so often lacking in airplanes powered with air-cooled engines.

According to the arrangement of the Curtiss Aero-

plane and Motor Company, the inverted engine, designated "Crusader," will be available to manufacturers for original installations during the remainder of the year and the engine will be in regular production in 1930.

The engine develops 130 hp at 1,800 r.p.m., has a bore of 4½ in. and a stroke of 3½ in., giving a total piston



A photograph of the inverted control stick installed on the Klean powered Blue Vindicator.

displacement of 441 cu. in. The overall length is 4 ft 8½ in. and the overall height 2 ft 8½ in. The weight is 345 lb.

An air scoop, located on the exhaust side of the engine deflects air from the upstream toward the cylinders. This device is proportioned in such a manner as to maintain the proper cylinder temperatures.

Each cylinder is made up of a forged steel sleeve machined all over and secured and shrunk into an aluminum head. The head is designed with integral cooling fins and has one intake and one exhaust valve. Cylinders are attached to the crankcase by clamps instead of the numerous bolt down studs.

A dry sump system of lubrication is used and, when the engine is in operation, the pistons are constantly throwing oil upward into the crankcase. Pistons have two compression and three oil rings. An "underhead" cam shaft, which is similar in action to the general type of overhead cam shaft, now in use on automobile engines, is included in the design. Accessories include two Scoville magnets for dual ignition and a Scoville governor.

The eight cylinder opposed engine exhibited by the Dayton Wright Engine Company, and known as the Dayton Green, is rated at 225 hp. The two banks of four cylinders each are set at an angle of 180 deg.

Several unusual design features were found in the engine exhibited by General Aeromarine Company. This power plant is a five-cylinder radial type having a rated output of 166 hp at 1,750 r.p.m. The weight of the engine is 355 lb.

The only geared engine at the show were Pratt & Whitney Wasp and Hornet of the Series C, which, together with the direct drive models at these engines, were displayed. These geared engines have been suitable for some use and probably will be used increasingly by airplane designers.

In the Wright exhibit were found all three members of the J-6 New Warhawk series as well as the Gypsy engine which is now being built by the Wright company.

One of the promised additions to the line of the Lycoming Manufacturing Company was exhibited, together with the initial product of the concern, which is now supported by the Department of Commerce. The rating given the new cylinder engine is 185 hp at 2,000 r.p.m. and the total displacement 645 cu. in.

The more recent seven cylinder engine is identical in construction to the older design and has a displacement of 800 cu. in. In both models the exhaust manifold is mounted between propeller and cylinders and is screw-driven.

The series D engine developed by the Auelson Aircraft Engine Company was shown and a number of refinements have been embodied in the design. Changes



A cloth cover under which is the engine of the new Blue Vindicator.

have been made in crankcase, cylinders and valve mechanism. A three piece crankcase has been substituted for the four-piece type. The cylinder head has been redesigned and has now been machined instead of being pressed on. Cylinder heads are secured and shrunk to the barrels and the intake port has been moved to the side. Compressing mechanism for the valves has been modified and strengthening of the rocker arms improved.

Three engines, including the new "Story-an," were displayed by the LeBlond Aircraft Engine Corporation. The Story-an is similar to the "Story," excepting the bore which is 1½ in. larger. The valve mechanism also has been modified. Although the engine was completed just in time for the show, it is expected that it will develop greater power and have a lower specific weight than the LeBlond Story.

The other engines exhibited are designs with which we are already familiar. These include the American Gypsy, Blue Jupiter, Broadback, Comet, Quack Challenge, Dayton Bear, Klean, Tips & South Super-Klean, Siskelly and Warner.

Two superchargers displayed bore out the prediction that greater use is to be made of these devices. One of these was built on the Roots principle, while the other was of the turbo type.

The Root type was shown by the American Cetus Engine Company and has been designed for use with the engine produced by that concern. It was stated that engines equipped with these superchargers probably would be used on the twin engine amphibian exhibited by the Great Lakes Vercelli Corp.

Mechanism for the built in turbo type was exhibited by the General Electric Company. This device is in the

Table 1—General Characteristics of a group of the airplanes exhibited at the Cleveland Show.

Manufacturer	Designation	Type	Engine	No. of Seats	Dimensions		Performance (FT)		Weights (Lb.)	
					Wing	Length	Max. Speed	Range	Empty	Loaded
American Klean	AKC-20	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
Albatross Hawk	ALB-1	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
Albatross Hawk	ALB-2	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-1	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-2	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-3	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-4	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-5	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-6	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-7	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-8	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-9	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-10	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-11	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-12	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-13	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-14	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-15	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-16	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-17	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-18	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-19	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-20	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-21	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-22	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-23	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-24	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-25	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-26	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-27	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-28	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-29	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-30	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-31	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-32	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-33	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-34	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-35	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-36	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-37	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-38	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-39	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-40	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-41	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-42	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-43	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-44	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-45	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-46	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-47	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-48	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-49	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-50	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-51	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-52	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-53	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-54	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-55	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-56	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-57	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-58	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-59	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-60	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-61	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-62	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-63	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-64	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-65	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-66	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-67	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-68	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-69	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-70	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-71	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-72	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-73	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-74	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-75	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-76	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-77	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-78	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-79	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-80	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-81	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-82	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-83	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-84	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-85	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-86	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-87	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-88	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-89	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-90	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-91	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-92	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-93	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-94	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-95	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-96	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-97	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-98	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-99	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000
American Eagle	AM-100	CH-300	6-cylinder	2	34.0	34.0	260	1,000	1,000	1,000

THE Airplane Design OF THE S. A. E.

THE SOCIETY of Automotive Engineers meeting with the Aeronautical Chamber of Commerce joining in sponsorship to discuss the theoretical problems of airplane design was held Tuesday morning as a part of the general aeronautical meeting of the Society. The keynote was devoted to the consideration of two lengthy and far reaching papers. "Wings—A Coordinated System of Their Design," by Ralph H. Upson, and "Criteria for the Structural Strength of Airplanes," by Professor Alfred S. Niles, formerly of the structural branch at Cook Field and now Professor of Aeronautics Engineering at Stanford University. Mr. C. H. Chatfield, chief of the aeronautical engineering faculty at the Worcester Polytechnic Institute of Technology and now with the Pratt & Whitney Aircraft Company, presided.

Mr. Upson's paper, which covered some thirty-five unemphasized pages and was strongly authenticated in spots, does not lend itself as a whole to easy abridgment or review. Readers interested in the detail of the author's procedure can only be referred to the original paper. It is, however, possible to summarize the objects, the principal methods employed and the important conclusions.

Mr. Upson's research was of a type that is often characterized by accumulated experience but that has never as far as the present writer is aware, been carried out with such thoroughness. It was his undertaking to secure the most general expressions possible for the weight of an airplane structure and for its total aerodynamic resistance, bearing in mind the most important characteristics of the members of the airplane as variables, and then by comparison to determine in general terms the most efficient wing section, plan form of wing, aspect ratio, etc., taking both weight and drag into account.

It has become common practice among those familiar with the several subdivisions of wing theory to divide the drag of an airplane into three parts, induced, wing profile, and parasite. Influence of the threefoldness of Mr. Upson's procedure is furnished by his having further subdivided the three foregoing as seven. Induced drag he left as usual under a single heading, since it is already fully covered by a mathematical formula, although he later made special comment on what he called "induced interference." Parasite resistance was split into three parts, the permanent and unchangeable portion originating elsewhere than in the wing structure, the contribution of the relievable bending or other variable wing bracing, and a variable portion due to such factors as the change of resistance of the fuselage with changing angle of attack. Profile drag was also divided into three parts, the basic amount due to skin friction, the added amount due to the thickness of the airfoil section and

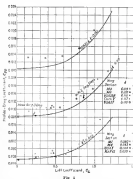


Fig. 1

the resultant increase in disturbance of the flow past it, and the additional profile drag due to the lift coefficient. Mr. Upson has found that the profile drag coefficient for good wing sections at zero lift exceeds the skin friction by an amount substantially proportional to the square of the thickness of the airfoil section (of course stated in terms of a fraction of the chord), and as the lift coefficient is increased the profile drag coefficient also increases by an amount in direct proportion to the cube of the lift. Curves bearing as the latter point, taken from Mr. Upson's paper, are presented as Fig. 1.

Weight was also divided into seven parts, one of them being the fixed load, including the metal load, power plant, and all parts of the structure not directly affected by the proportions of the wings. The other six were respectively the weights of the flanges of the wing spars, the webs of the spars, the structure of the ribs, the internal drag bracing, the external wing bracing and the covering or skin.

SESSION

By EDWARD P. WARNER

In making his analysis the author, having been concerned with the construction of the Aeronautical Chamber at the time of preparing the paper, took that machine as a basic example. It will be recalled that it is a low-wing monoplane with cantilever wings covered with plywood. The span is 40 ft., the wing area 196 sq. ft., giving an aspect ratio for the tapered wing of almost exactly eight. The gross weight is 1325 lb.

Space limitations prevent the reproduction of Mr. Upson's arguments in full, but a few salient figures may be drawn from their contents. Of the weight of 1325 lb., it was estimated that 1000 lb. represents the fixed load, the remainder coming from the variable portion of the wings. Of the 225 lb. remaining, the wing covering, which is entirely of plywood, constitutes almost exactly one-half. One-quarter of the total comes from the wing spar flanges, and one-eighth each from the ribs and the spar web members. There is of course no external bracing and the drag bracing is replaced by the rigid skin.

The basic set of weights is of course independent of flight conditions, but that of drag depends upon the lift coefficient, and Mr. Upson made the drag analysis for several different speeds. At 100 m.p.h., the approximate maximum speed of the airplane, the fixed portion of parasite resistance constitutes 129 lb., or 61 per cent of the total. The skin friction was 53 lb., and the additional profile drag due to thickness of the section, in spite of the unusual thickness necessary for the cantilever structure, was only 42 lb. The induced drag at this speed was 15 lb.

At 60 m.p.h., once the speed of best climb, the distribution changes very markedly. The total drag has dropped to 113 lb., and 42 lb., or more than one-third, comes from the induced drag, which constitutes less than ten per cent of the total at the highest speed. Fixed parasite drag still re-weighs the largest factor, however, with 45 lb., while skin friction, varying approximately in proportion to the square of the speed, has gone down to sixteen. The additional profile drag due to lift coefficient has begun to have some significance at 60 m.p.h., adding 7 lb. to the total.

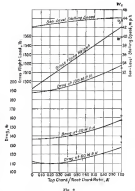
Wing Flange and Performance

ALL THIS was preliminary to Mr. Upson's real object, that of finding the total effect of changes in the design. His first step was to assign a constant area and aspect ratio and investigate the effect of changing the streamlines of the wings in plan form. There are of course various structural assumptions which might be attached to a change of taper. In this paper the author elected

to keep the same weight and dimensions of spar flanges throughout. Increasing the amount of taper, since it increased the chord of the wing at the root, therefore made it possible to reduce the relative thickness of the section.

The only important change in weight to the taper was caused in these assumptions, which seem very reasonable ones, came from the ribs and the spar webs. In both cases there was a marked saving with increasing taper, largely because of the relatively thinner sections used. It was found, for example, that the webs would weigh 53 lb. in a rectangular wing, while tapering to a tip chord only half that at the root was found to reduce the weight to 38 lb. and further tapering to a tip chord one-quarter of the maximum at the fuselage brought a further reduction of the figure to 30. The corresponding figures for the weight of the ribs were 58, 40 and 32 and for the total weight of the airplane 1361, 1330 and 1310.

Venomous in drag were more important. With a rectangular wing the total drag calculated for the airplane was 208 lb. at 100 m.p.h. and 127 at 60 m.p.h., figures which were reduced to 196 and 114 by tapering to a tip chord one-half that at the root and to 152 and 109 by further tapering to 25 per cent. Ninety per cent of this saving at the high speed, and about one-half at the lower velocity, resulted from the reduction of the profile drag as the result of the use of thinner wing sections with the apex tapered plan form. Practically all of the remainder appeared in the induced drag item, since a taper to a tip-chord of a little less than a half (set at the root) gives the closest approximation to the theoretically ideal distribution of lift along the span.



Mr. Upson's curves of total drag at various speeds and of weight are plotted against the amount of taper in Fig. 2.

A similar investigation was made on the effect of changing span while keeping a constant area. Increasing area of course increases profile drag slightly, since thicker sections must be used, and increases the structural weight considerably, but raises a great saving in induced drag, especially at cruising speeds. The general results are obtained by Mr. Upson are reproduced in Fig. 3, where it will be noted that the drag remains almost constant at high speeds, since the induced portion is there relatively unimportant but varies sharply with span at 60 m.p.h.

Results of a similar investigation on the effect of changing area, while keeping a constant span (the aspect ratio therefore being increased with decreasing area)

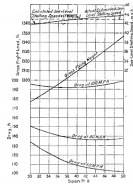


Fig. 1

are displayed in Fig. 4, drawn like all the others from the original paper. It again proves that the variation in weight are relatively more important than those in drag.

For the study of the effect of changing thickness in air the two cases last mentioned are given. Mr. Upson gave two detailed figures for drag and weight by separate means. Again, therefore, the data are presented simply in graphical form in Fig. 5. It was a little surprising at first sight to be told that an increase of wing thickness beyond fifteen per cent of the chord would yield no further saving of weight in a taper-wing straight cantilever structure.

In other words, beyond that point the weight of rib

and spar ribs was found to increase more rapidly than the weight of the flanges could be diminished.

External Bracing or Cantilever Wings

THE EFFECT of introducing external wing bracing was investigated by direct comparison of the efficiency of a rectangular cantilever wing and that of the same wing with two different sets of wing bracing. In the first externally braced case the only added members were struts running directly from the top logarithm to a point of attachment half way out on the upper surface of the wings (since the machine was of the low-wing type). The second alternative was to provide these struts with intermediate bracing by light reinforcing members attached at their middle points and connecting to the wing.

The external bracing made it possible to reduce the wing thickness from 21 per cent to 8.8 per cent of the chord. The effects as tabulated by Mr. Upson were surprisingly small, neither the drag nor the gross weight of the plane being changed by more than 2 per cent.

In general, the external bracing without intermediate reinforcements showed itself slightly worse than the rectangular cantilever arrangement, whereas that with the intermediate bracing for the struts was slightly better. Both were distinctly inferior to the tapered cantilever wings. Better results might have been obtained, however, if the externally braced wing had been made a little thicker and given lighter spar flanges instead of being retained in thickness in direct proportion to the reduction of bending moment and having the flanges kept the same in weight.

In the paper in print, Mr. Upson went on to emphasize the importance of interference between the fuselage and the wing and especially of what he called, as according to a suggestion by Mr. J. D. North the induced interference. The effect of a change of tail aerodynamic form in changing the effective working aspect ratio of a wing, especially as a low-wing monoplane, may nearly double the induced drag calculated by ordinary formulas. Data from a test of a Jenkins machine was cited in support of this statement. Figures taken from Nomenclature Advisory Committee tests at a Sperry Messenger recently show that for a biplane the direct or so-called interference between the wings and other parts of the structure may add as much as 60 per cent to the parasite drag of the fuselage and leading spar used alone. The author closed the written paper with the general conclusions stated as follows: "A set of externally braced wings of ideal proportions has from 30 to 30 per cent less drag, exclusive of that due to the fuselage, and slightly less weight than the best arrangement of rectangular wings either externally or internally braced except in a case where bracing with wire alone is possible. Considering also the improved stability and control afforded by tapered wings, the importance of devising economical production methods for such airplanes is apparent."

Adequate span is the prime concern for any design with a large power loading. Many small airplanes have been failures partly from this cause. In summarizing the paper verbally, however, Mr. Upson went on to give some additional data not included in the printed copy, supplementing his discussion of his own work with a vigorous plea for the more general use of mathematical methods in airfoil section design and selection. He presented a curve of maximum lift against mean camber of airfoil section which is approximately reproduced in Fig. 6. Individual sections of course vary in their properties but in general, for

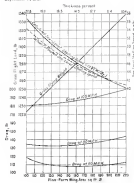


Fig. 6

smooth and well-designed lines, the maximum lift is a function of the mean camber.

Certain conclusions were drawn about the wing section which should be selected on the assumption that the ratio

aspect ratio
thickness ratio

remains constant, presenting that the span of a wing of constant chord could be doubled without sacrifice of efficiency if the thickness were also doubled. This may be questioned, as the bending moment in a spar varies in proportion to the square of the span and the strength in bending for a given spar weight varies only as the first power of the thickness. Taking the assumptions as stated, however, Mr. Upson concluded that for a span of fifty times the thickness of the wing a maximum lifting ratio of 25 would be obtained by using a thickness of twenty per cent of the chord, with an aspect ratio of ten.

Shortage of time prevented any extended discussion of Mr. Upson's paper, and the meeting turned at once to the consideration of Professor Niles' work on load factors and load distribution. Since the author was unable to attend the meeting, his paper was presented in abstract by the writer of this review.

Stress Analysis and Load Factors

THE PAPER concerned itself especially with the present rules at the Department of Commerce for loading and stress analysis of airplanes and the possible desirability of modifying them in various respects, and the discussion was divided into three or four distinct parts. The author disposed rather summarily of the possi-

bility of a general revision in stress analysis procedure which would provide for the design of structures to meet the maximum stress in any member under the most severe anticipated working conditions not in excess of a definite fraction of the yield-point for the material, instead of following the present standard practice of designing to carry approximately one and a half to two times the maximum anticipated load before there is actual collapse of the structure. The factor of 1.5 or 2 is, of course, the true factor of safety to allow for deterioration of the structure, or for imperfections of material or workmanship. There has been feeling among many students of structural theory, especially in England, that to base analysis on working stresses before the yield-point would be more logical than the present method, but Professor Niles expressed the view that a quite sufficient objection to the change would be the impossibility of checking the stress analysis by a static test to destruction. Only if the analysis is based on the actual failing load of the structure will a static test be useful for check.

A less academic question, and one to which the author devoted much attention, is the possible desirability of modifying the load factors now specified for commercial airplanes. Professor Niles tabulated the results of all the aerodynamic tests that have been made in recent years at McCook and Langley Fields, and placed beside them the load factors for which the rules would now require that the various planes used in the tests should be designed. On the PW-3, for example, load factors running up to 3.5 have actually been experienced in flight (figures even considerably higher than that have been assumed in more recent tests at Langley Field), while the Department of Commerce rules would require designing such a machine for a factor of 2.7, leaving a margin of safety of only five per cent. In two of the

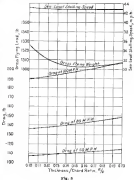


Fig. 7

eighteen airplanes unloaded by Professor Niles as having had the subject of accelerometer tests the margin of safety was less than the 50 per cent which ought to be considered as the minimum. The conclusion which prompts me to be safe were in most cases of large size and relatively slow. For present planes, the present factors appeared uniformly too low.

Assembling these data at some length, with incidental reference to the probable magnification of loads experienced in bumps and air and to our lack of adequate information on that subject, Professor Niles concluded that the present practice of classifying planes by weight and power loading was as reasonable as any that could be adopted unless a classification in terms of type of service could be arrived at, a process of admitting difficulty, and that "The load factors called for by the Department of Commerce are sufficiently high but not too high for heavy and lightly powered designs, the load factors for the lighter and more heavily powered designs should be raised so that the severe maneuvers to which these types are subjected, particularly in landing, will not cause excessive stresses." The paper especially recommended the present practice of calling for twenty per cent more strength in brackets and joints and fittings in general than is required in the major structural members.

The second general topic raised was that of load distribution on the structure. Professor Niles felt that

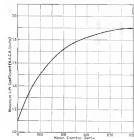


FIG. 1

the scarcity of data on pressure distribution prevented any confident recommendation of changes in loading at the present time. He did, however, suggest that the maximum intensity of loading on the wing tip is probably two-thirds as potent, and that the bending moment at the outer strut point in biplanes ought to be further increased although it is at present hardly possible to give any accurate general rules for the correction to be made.

One loading condition of which the immediate review was strongly urged was that occurring the normal dive. Professor Niles pointed out that he had had a hand in the original preparation of the present specifica-

tion on that point, and that it had been planned for tentative and temporary use. It is especially defective, he said, in that it makes insufficient allowance for differences between the various aerial sections that are used, some of them imposing much more twisting action on the wing truss than others during a dive. He urged that it be fundamentally reconsidered.

Designers frequently complain that the Department of Commerce regulations on stressing are too involved and require the examination of too many load conditions. The author expressed but little sympathy with the first complaint, much more with the second. He believed that some combining of loadings to reduce the total number of conditions calling for separate treatment should be possible, especially in the landing and take-off portions. He recommended that any change of the sort be made only after long and careful deliberation and study of the effect on specific cases.

Static Testing

THE LAST GENERAL POINT considered here upon the desirability of static testing of commercial airplanes. On this subject Professor Niles expressed no revolution, any conclusions. He suggested that for some parts, of which plywood ribs furnish a good example, static tests are much easier and cheaper to make than reliable stress analyses, and should accordingly be directly dispensed with. Other structural elements, of which welded steel tube fuselages were given as an instance, have been found by experience to be subject to such reliable and accurate stress analyses that static tests are quite unnecessary. In any case, the author suggested that static tests need be employed only for a restricted number of loading conditions, the results being extended to cover other cases by mathematical analyses. Although he was not disposed to urge going to extremes in static testing, he did feel that a load to destroy the part is a general check on any design about to go into quantity production, for "Stress can always creep into a stress analysis, and accidents may be made."

Professor Niles urged that more careful attention be given to stressing in general, using planes existing in their very modernity. "The work," he said, "does not regard it a hardship on the constructor to ask for either a static test or a stress analysis of every main structural part of an airplane. The designer should have such data for his own information before offering his design to the public. Equally too great a handicap exists to design either by part guess work, empirically applied 'general experience' or by blind copying, in which engineers take the place of the state rule, without checking the results by any adequate method. In the end each design should be backed by tests made either on the design in question or on similar designs."

Lamentations of late again checked discussion. The only extended remarks were made by the present writer, who urged that it was quite impractical, as shown by all accelerometer tests made, to design an airplane which would be incapable of being broken in the air by a heavy-handed pilot, and that it should therefore be frankly recognized that a certain degree of discretion would be observed by the pilot and that certain limitations would be put on his use of the machine. He suggested that if adequate notes were posted on the instrument board, the pilot must not be dared to exceed a certain speed, the designer would be justified in assuming that these instructions would be at least approximately followed, and in choosing his load factors accordingly.

THE AERONAUTICAL CHAMBER

Talks of Many Things

Department of Commerce Regulations, Standard Fuels, Uniform Discounts, and Production Reports Among Subjects Discussed at Cleveland

By EDWARD P. WARNER

THE MEMBERSHIP of the Aeronautical Chamber of Commerce came to Cleveland with so much business to do that they followed the practice, established on several other recent occasions of meeting always in sections and never as a whole.

Since the most crowded program was that for the airplane manufacturers, alone or in consultation with other groups, they got away to an early start on Monday morning under the chairmanship of J. D. Thompson, president of the engine bearing business. The Commercial Aircraft Manufacturers' Section, according to a list corrected to August 7, numbered 63 members. Only about twenty-five individual representatives of the industry, however, attended the opening meeting, where the program included topics of far-reaching importance to the whole industry.

The Production Report

AN OUTSTANDING feature was the presentation by Charles E. Parker, of the Chamber's Staff, of the report of airplane production for the first six months of the present year. A substantial part of the report is printed elsewhere in the current issue. Its most striking features, apart from the overall total of 2,844 commercial and 327 military airplanes produced during the six months by firms furnishing reports to the Chamber, were the virtual disappearance of the cabin biplane, the enormous growth in production of cabin monoplanes in general and multi-engine monoplanes in particular and the increasing popularity of flying boats and amphibians. Of the total estimated production of 3,900 planes, approximately 90 per cent came from the plants of members of the Aeronautical Chamber.

The production of open-cockpit biplanes for the first six months of 1939 was approximately two-thirds of that for the whole of 1938. The number of cabin monoplanes built, however, exceeded the figure for the entire previous year, and the number of multi-engine ma-

chines was almost double the 1938 total, as were the production both of flying boats and of amphibians.

The report showed prices remaining relatively steady, with a slight upward tendency. Roughly speaking (the report contained no basis for an exact calculation for individual types) the retail value of the open biplanes without engines averaged \$3,000 during last year and \$3,800 for the first half of the present year, while the corresponding figure for cabin monoplanes was \$6,800 and \$7,200. The flying boats and amphibians, all lumped in together, increased their average from \$12,300 to \$25,150. The aggregate commercial production for the six-month period was \$127,087,497, not including engines, and the military construction contributed \$6,552,825 more. The analysis of the engine figures, later given in a separate report, showed that the industry's production during the first half of 1939 was almost exactly 75 per cent commercial.

The report on engine production was not released until Tuesday morning, when it was laid before a meeting of the engine manufacturers, but it is best to report it immediately beside the airplane figures. The production of commercial engines reported by fifteen different manufacturers totaled 3,273, or about 15 per cent in excess of the airplane figures. Allowing for the number of multi-engine airplanes built, however, the actual excess of engines was reduced to about 4 per cent. For the previous year the number of commercial engines built had been almost exactly one-half of the number used in new airplanes. The influence of the new-type engine over that period, and its extraordinary rapid decline in the last six months, stands out from these figures too clearly to need discussion. More engine bearing as the same point is furnished by the break-down of engine production into classes by power output. In 1938 the reported production of engines of less than 200 hp was about 1,000, totaling only 833 units. For the first six months of 1939, with the C-45 definitely in the

background, the corresponding total has jumped to 1,550. Production in the 200-300-hp. class fell off considerably, but it is easy to rally that this was due to the cessation of production of the Wright J-5 during the early months of the present year, that power plant having been the mainstay of the 200-hp. division during 1928. Another striking point, symbolizing the increasing scope of the "luxury ship" with high power and loss of speed was the increase from 475 to 1,342 of the number of commercial engines of over 300 hp.

The total retail value of the engines built in the period covered by the report was \$10,500,000 approximately, as against \$8,037,900 for 1928. Military production increased the aggregate business done by the engine industries to something over \$24,000,000 in this month.

Retail prices for last year and this year cannot be compared, as no detailed break-down of the value of output was made for 1928. For the more recent period, however, separate notes were compiled and show a remarkably close adherence all the way along the power scale from 50 hp. up to 450, to an average retail price of \$13.00 per horsepower. Lining out of consideration the engines of less than 50 hp., on which the average would be expected to run higher, our group rose to \$17.00 per horsepower, but all the rest hung very close to the \$13.00 mark. These averages, again, are unofficial deductions by the writer from the data contained in the report.

With the production report before them as an evidence of the magnitude of their operations and of their merchandising problems, the manufacturers proceeded to details. The topics of liveliest general interest were those concerning the relations of the industry with the Department of Commerce, discussed in satisfaction of the meeting with the Department's representatives later in the week.

Administrative Regulations

IT WAS REMARKED that the general desire was for uniformity in the Department's practices and procedure, rather than that the rules should take any particular form. Various manufacturers expressed themselves as working to be safeguarded either against sudden changes in administrative regulations which would upset production schedules or against arbitrary enforcement of personal ideas by individual inspectors. The code of procedure previously outlined by the Department of Commerce was discussed and was in general approved. It provides for the publication of administrative regulations once each year, on January 1, and for an open conference of the Department officials with representatives of the industry preliminary to each revision. The manufacturers' meeting recommended by resolution that changes in the requirements should be definitely announced six months before becoming effective, to give time to get ready for embodying their results in current designs.

Some of the manufacturers protested that individual inspectors had tried to force the adoption of their own ideas of good design and expressed a desire for an inspection board which would set forth in detail the scope of the inspector's authority and the manner in which he was to exercise it, serving both as an instruction book for the inspector himself and as a guide to assist the manufacturer what he might anticipate. Mr. Paul H. Brittain, Washington representative of the Chamber, announced that the preparation of such a manual was in early progress.

Among the detailed regulations, now in existence or suggested for early adoption by the Department of Commerce, the strongest feeling was manifested on the proposed rules to guard against over-speeding of engines by the use of small or low-rpm propellers and on those covering the stability and seating requirements. There was evidence of a general feeling that freedom from liability to fall into a trap was more important to safety than ability to remain in for a long time and there consent quickly, especially in those cases in which overhaul



Left: Alex. Brown, E. A. Davis, Mr. George Elmer. J. Lee Alexander

and abnormal interpretations of the controls might be necessary in order to force the machine to quit. There was a clear consensus among those who could themselves point out that it should in any event be the uniform and permanent policy of the Department of Commerce to prescribe limits and not methods, and that, to take a concrete case, if an airplane had adequate stability it should be no concern of the regulating authority whether an adjustable stabilizer or some other device was used as an aid to securing it.

Opposition to the proposed regulation prohibiting the use of propellers which would permit the engine to run more than 2 per cent beyond its rated speed was conspicuous from several quarters. Objection was made especially as the grounds of difficulty of control and enforcement.

A committee of three was selected to interpret the collective views of the industry and take the lead in presenting them to the officials of the Department of Commerce. The committee included L. V. Kerber, formerly head of the engineering section in the Department's Aeronautics Branch but now president of the Sperry Company, Captain H. C. Richardson, lately in charge of aeronautical material for the Navy and now chief engineer for The Great Lakes Aircraft Corporation, and Mr. A. W. Mooney, president of the Mooney Aircraft Corporation.

A committee of the Chamber has been at work on the study of a standard accounting procedure for aircraft manufacturers and has decided a questionnaire among all those interested. The results were presented at the commercial manufacturers' meeting in the form of a first draft of the proposed standard procedure, including standard forms for sales records, purchase requisitions, cost keeping in manufacturing processes, etc.

A point on which there appeared to be general agreement was that the number of aircraft shown should be

rigorously kept down, and that so far as possible they should be held where there would be no competition from other structures.

The chairman told the meeting of the progress that had already been made in developing a standard sales agreement for the use of exporters in determining their relations with their foreign distributors. To further aid the exporter a committee is now at work on the development of a general table code for the special use of the aeronautical industry. Assistance has been volunteered from the automobile industry and other quarters where such codes are already in use, and one of the manufacturers at the meeting offered the code already in use by his own company as a foundation for the Committee's work.

The directory of sources of supply for materials and parts which has been the subject of discussion at previous meetings of the Chamber was presented at the Cleveland session. It includes at present a listing of 80 distinct principal products used in plane manufacture.

In closing the airplane manufacturers' meeting Mr. Alexander spoke of the success attained by the glider movement, especially in schools and colleges, and urged upon the manufacturers of planes that they do everything possible to aid it as one of the best means of spreading aeronautical enthusiasm and encouraging and assisting potential pilots and airplane owners of the future.

Discounts and Service

BORNE at the airplane and engine manufacturers' meetings there was discussion of the question of uniform discounts on engines, of the relation between the list price of engines and the price to the airplane manufacturer, and of the way in which the servicing of engines should be handled. There seemed to be no uniformity

of opinion on either subject, and the view was several times expressed that it was too early for the adoption of a standardized practice. In view of the complexity of the subject a special committee was appointed to study it, with particular reference to the experience of the automobile industry. The committee includes five representatives of each group.

Upon servicing, however, the majority of the airplane and engine people concurred that in view of the spread shop equipment and training of the personnel required, the servicing of engines should be left in the hands of the engine manufacturers, to be accompanied through their local distribution centers. Another topic of lively interest to the engine manufacturers was the method of having engines for approved type certificates. There were some individual expressions, but no apparent consent of opinion, that the present rules should be liberalized to permit the making of type tests at the engine manufacturer's own plant, instead of only at the Bureau of Standards or other Government laboratories.

Representatives of the engine industry spoke with particular force of their need for a better break-down of the present market, with analysis into its several parts on various bases of classification. Recent was expressed that statistical information of the type provided by the Chamber's production survey was not more generally and consistently available and in such more detailed form, and the representatives of certain manufacturers spoke of their own efforts to keep their statistical records complete and up to date and of their desire to co-operate in obtaining such data for the industry as a whole.

Meeting with the Department of Commerce

THE MEETING on Friday between the airplane manufacturers and the representatives of the Department of Commerce was presided over by the Hon. William F.



An immense crowd in the quadrangle at the Cleveland Airport during the Air Show.

MacCracken, Jr., retiring Assistant Secretary of Commerce for Aeronautics. It was ideally harmonious, the most controversial points having been discussed and new compromises reached in preliminary discussion between the department's officials and the committee. It was so generally agreed that the committee should be so organized as to represent the industry. It was suggested that the plan for protecting producers against sudden changes in airworthiness requirements be modified somewhat from the original draft and that, as previously suggested, the manufacturers' meetings should be a definite interval between the date of announcement of a change and the time of making it effective. The committee report as presented to the meeting proposed that changes in the regulations might be issued at any time during the year but that they should be made at a fairly regular interval. Meetings of the Department of Commerce noted that four months after the issuance of the announcement. It was further provided that when an airplane design has once been approved no subsequent revision in rules would affect the construction of airplanes from that design. The definite assurance of such a procedure would be an invaluable concession on the part of the Department of Commerce, which would thereby lose the power to require that an existing design be modified, even after a lapse of a number of months, in case an accident or other factor of danger which might have required modification of the airworthiness rules was subsequently found to be a design's first appearance.

It was further recommended by the meeting that changes in airworthiness requirements might be suggested by any manufacturer at any time, and that recommendations submitted either by manufacturers or by the Department of Commerce itself should be considered by the industry for comment during the usual period before the change would become fully effective on new designs.

A further proposal upon general procedure was that in case of controversy between the Department of Commerce and an individual manufacturer the approved manufacturer should have the right to appeal to the chairman of the Air Council of the Chamber to appoint a committee to study the matter. If the committee considered such an appeal justified it would be able to throw its own weight and influence into that of the whole industry in making a request to the Assistant Secretary of Commerce for joint discussion and review of his department's previous decision. Secretary MacCracken expressed approval in principle of these proposed revisions, although there was no definite consensus on the part of the Department of Commerce.

Spinning Trials

SPINNING was discussed along with the same lines as the other commercial manufacturers' own meeting. Capt. H. C. Richardson, and E. B. C. Nordmark of the Bellanca Company, both indicated their belief that the emphasis should be laid upon the desirability of cutting out spinning entirely, first, by eliminating the possibility of spinning from among the design characteristics of the airplane and secondly by discouraging pilots from carrying out any circumstances the prolonged continuance of an spin which might inadvertently get started. Mr. Lawrence, vice-president of the Curtiss-Wright Corporation had a different view. He believed that there were few pilots who would in any case prudently spin any machine capable of performing the maneuver and he thought that there should be no severe restrictions on the test and spinning it was necessary. The

present rules of the Department of Commerce recognize the probability of the prolonged spin. One or two other speakers at the meeting agreed with Mr. Lawrence, and even suggested that power spins should be included as a part of the test of new planes.

There was objection voiced against the overwrought limitations on airplane engines and propellers and against the present method of testing landing gear by static drop, but no final decisions were reached. Secretary MacCracken said that he intended to take the discussion under advisement as a basis for their further decisions. "Through the friendly relation of the Department of Commerce and the aircraft industry's trade association, the Aeronautical Chamber of Commerce," Mr. MacCracken had said to the assembled manufacturers in closing the chair, "we have been able to discuss the problems that protect the public and meet the desires of manufacturers." The desire to insure the continued accomplishment of both these ends, without having to sacrifice the interests either of the public or the industry, plainly animated both the governmental and industrial groups at the conference.

Standard Fuels

THE treatment of the petroleum industry in aeronautical problems was reviewed at the meeting on fuels and lubricants, where eleven different oil companies were represented, along with several airplane engine manufacturers and the Government's interests. The spirit of the meeting appeared to be favorable to the development of standard specifications, so that pilots moving about the country might buy fuel and lubricants by grade instead of having to depend upon a trade name or oil of which might be geographically restricted. Standard was also emphasized in favor of the production of fuels of higher quality, suitable for developing the full capacity of high-compression engines. After preliminary discussion among themselves and with representatives of the petroleum industry, the fuel and oil engineers held a meeting with the air transport operators for the discussion of fuel standardization and of the grades acceptable, as well as for the study of fueling equipment and methods of servicing at airports and the possible interest of modifying them when standardization was complete. The transport operators and the representatives of the oil companies found it easy to agree that there was room for marked improvement in fueling methods and reduction in the time consumed by the operation in most cases.

The fuel companies' representatives collectively viewed with some alarm the occasional practice of giving exclusive concessions for the supply of fuel and lubricant on public airports to a single company. They formally recommended that all airports be left open to the use of any supplies of fuel and of oil willing to meet with reasonable regulations and put in their own equipment.

Mr. E. E. Alden of the Standard Oil Company of New Jersey was chosen chairman of the fuel and lubricant section for the coming year.

The studies of economics and materials, like the leaders of airplanes, turned at some length on show policy and drafted to formulate a questionnaire on that point among their membership. It was voted by their general that membership in the accessory manufacturers' section should be limited to the makers of such equipment as appeared in the standard airplane including periodicals furnished by the pilot for eliminating manufacturers of airport equipment and other ground accessories. The

question of limitation of membership, with the object of securing full responsibility on the part of all members and of keeping the Chamber's program a mark of distinction, had also been discussed by the commercial airplane builders. When the leaders of the definition of discipline adopted by the accessory manufacturers, 27 concerns belonging with their section were represented at the accessory meeting. The members agreed on the importance of furthering the use of air transportation, and pledged themselves in such a campaign to promote travel by air. James R. Fitzpatrick, vice-president of the Haskelite Company, was chosen as permanent chairman of the section.

While the airplane and engine manufacturers were chosen to form one room and the fuel experts in another, the distributors and dealers were conducting a session of their own, largely for purposes of organization of their newly formed section. George West of New York was appointed the chairman of the executive committee, which contains six other members representing as many geographical sections.

The materials of the merchandising problem was impressed upon the meeting with due solemnity and the manufacturers, through Mr. Alexander, pledged their support to the co-operative work that the dealers and distributors might undertake and the organization that they might form. It was agreed that the leading up of adequate sales organizations, covering the ground completely but without undue overlap, was a major if not the dominant problem of the aircraft industry at the present time.

About the desirability of encouraging personal travel by air, the distributors and dealers were in hearty accord with the manufacturers of accessories, and they too promised their support to a carefully planned and extensive campaign to encourage the use of airlines by refraining the public upon their usefulness. The view was firmly expressed that travel upon established routes was an accessory industry, dependent by the character of private flying and private ownership upon a good sale.

Flying School Regulations

ON THURSDAY, flying schools came up for discussion at a special session, and it was made evident that the aeronautical industry was united in its attitude toward the Commission, and the Aeronautical Branch of the Department of Commerce was going to keep after the "fly-by" flying school menace until it is entirely and permanently removed.

Colloquial attention was first given to the problems of flight instruction and the flying school at the conference of the industry with the Aeronautical Branch held in Washington in December, 1927. After considerable argument as to how much restriction a student should have before being sent solo, how many solo hours that student should put in on his own before being granted a license, what equipment a school should have, and many other items, it was decided to let the schools and "so called schools" endeavor to solve their own problems during 1928. If by the end of that time matters were not on a satisfactory basis, the industry and the Branch would step in and take a hand.

Unfortunately, the end of 1928 found some spots still existing, and during the Chicago Show a flying school conference was appointed by the Aero Chamber. This conference, working in cooperation with the Aeronautical Branch staff, has since then been endeavoring to bring the rising of flying schools throughout the country by the Department of Commerce. In the meantime, special

legislation introduced and passed in passage by Senator Bagley enlarged the Department's powers in that field.

At the meeting in Cleveland last week the industry-Aeronautics Branch conference took another step forward in the solution of these problems. The national flying school committee, composed of C. S. Jones of the Curtiss Flying Service, chairman, Oliver Parks of St. Louis, C. C. Mason of Washington, W. D. Haviland of St. Louis, John Rogers of Los Angeles, Jack Frye of Los Angeles, and other aviation men are members. The national flying school committee, composed of C. S. Jones of the Curtiss Flying Service, chairman, Oliver Parks of St. Louis, C. C. Mason of Washington, W. D. Haviland of St. Louis, John Rogers of Los Angeles, Jack Frye of Los Angeles, and other aviation men are members. The national flying school committee, composed of C. S. Jones of the Curtiss Flying Service, chairman, Oliver Parks of St. Louis, C. C. Mason of Washington, W. D. Haviland of St. Louis, John Rogers of Los Angeles, Jack Frye of Los Angeles, and other aviation men are members. The national flying school committee, composed of C. S. Jones of the Curtiss Flying Service, chairman, Oliver Parks of St. Louis, C. C. Mason of Washington, W. D. Haviland of St. Louis, John Rogers of Los Angeles, Jack Frye of Los Angeles, and other aviation men are members.

The shifting committee of the Chamber would also help by sending to newspapers and magazine copies of the Chamber's reports with the request that advertising of fraudulent schools and those with grossly inadequate equipment be refused publication.

The Cleveland meeting was presided over by Clarence M. Young, newly appointed Assistant Secretary of Commerce for Aviation. In his introductory address Mr. Young said of the Department's activity in the matter of protecting prospective pilots against such schools as the Department has been unable to control by its regulatory jurisdiction, "we cannot go beyond this brief. Therefore it remains for the Aero Chamber to take drastic and vigorous action in the drive against questionable flying schools."

The Senate and Interstate Section meeting was devoted largely to the further consideration of problems discussed at a meeting held at the Chamber's Washington headquarters a few months ago. Edward Howard, chief of the Regulations Division of the Department of Commerce, again explained the inadequacy of the present system of recording laws and concepts on the map of the United States. Under the Uniform State Aeronautics Law, which is in force in eleven states, the owner of a plane is liable for all damages caused by his plane, whether through his negligence or not, and under federal law, the owner of a plane is also responsible for payment of fines for violations of regulations.

It is therefore necessary to protect the insurance companies against their being responsible for damages caused against a plane of which they are financing the purchase, and of which they therefore stand on the outside of the case. It is also necessary to protect the insurance companies against their being responsible for damages caused against a plane of which they are financing the purchase, and of which they therefore stand on the outside of the case. It is also necessary to protect the insurance companies against their being responsible for damages caused against a plane of which they are financing the purchase, and of which they therefore stand on the outside of the case.

"According to various representatives of the insurance industry, the present aeronautics laws in aviation have to be based upon a realistic and comprehensive basis available on rules necessary." They recommended that steps be taken to obtain more definite and complete data on rules, for in this direction by the hope of lowering aviation insurance rates. It was suggested also that there should be a more extensive exchange of personal and credit information between the insurance companies and the industry, and that there should be a more extensive exchange of personal and credit information between the insurance companies and the industry, and that there should be a more extensive exchange of personal and credit information between the insurance companies and the industry.

THE National Air Races, AND IN SUMMARY

DAY BY DAY

By JOHN T. NEVILL

SIFTING new and wonderful precedents from nearly every angle, the 1929 National Air Races, held at Cleveland from August 24 to September 2, set a new standard which every subsequent meet will take as their goal.

True, the same thing was said, and very properly said, of the Los Angeles meet last year, but the Cleveland Chase, by sheer magnificence, diversity and completeness of character, and smoothness of operation has created a new pattern for outdoor aeronautical Olympiads.

It is always easy to find ground for criticism for any

rent of this sort, and it always will be until the human element is removed entirely, but in Cleveland, Managing Director Cliff Lindbergh, Air Race Chairman Floyd J. Logan, President Mrs. Bradley, Vice President L. W. Greve and the countless others who had a part in the Races, asking, put on a show far exceeding, in nearly every way, what was to be expected of a reasonably expert.

The outstanding factor in the 1929 National Air Races was the success achieved by the management in injecting nearly every phase of aeronautical activity somewhere into the program. Monoplanes, biplanes, and seaplanes,

planes, biplanes and amphibians, single-engined planes and multi-engined military, naval, and commercial planes, open cockpit planes and those with cabins, high-powered and low-powered machines, autogiros, gliders, fabric-covered and metal-clad and rigid airships, men and women contestants, derbies from the east and west, speed contests and acrobatics by commercial planes and speed contests and acrobatics by military craft—all these, and more, were on the bill.

Taken as a whole, the daily program of racing and stunt flying clearly revealed the progress made in airplane performance capabilities during the past year, and pointed also to the growing strength and technical capacity of an industry which is beginning to turn out specialized racing machines on its own account.

Although a detailed report of the events of the first Sunday and Monday were contained in last week's issue of AVIATION, for the sake of continuity they are tabulated herewith, with some additional comments.

THE first event, closed-course racing, was a circular 30-mile 10-lap race for planes powered with OX-5 engines. It was called off, as some pilots flew the five-mile and others the ten-mile course. The remarkable difference in racing skill among pilots of unquestionable general ability became apparent in the very first lap of the first race, and continued a striking feature throughout the week. Some of them, but irregularly few then, just wide enough as they approached the pylons, started their bank at just the right time, and swept around the turning angle with but a few feet to spare. Others made every conceivable error, the commonest fault being to fly directly at the pylons and start the turn too late, swinging far beyond the mark, as much as two or three hundred yards in some cases, before the turn was complete. Some pilots, too, especially those flying older machines with but little reserve of power, then very low and then lowering wings about turning their planes up on their wing tips, edging around the turns with less than thirty or forty degrees of bank. In the shorter OX-5 contest the turns made by Douglas Davis in a Travel Air were especially notable for their smoothness and closeness to the pylons.

Event No. 23, next on the program, was a 120-mile 12-lap race for National Guard planes powered with Liberty engines. Maj. John S. Owens took first at 149.95 m.p.h. Second was Lieut. T. E. Baker, who had flown in from Boston, 146.4 m.p.h. Third, Lieut. P. R. Love, well known as Colonel Lindbergh's former flying mate, 144.58 m.p.h.

The arrival of the women's derby from Santa Monica followed Monday afternoon. First was Mrs. Lenora



Colonel Lindbergh landing one of the "Whit" planes in wheely position.

McPhetridge. Thelma in 26 hours, 19 min., 2 sec. flying time. Second, Gladys O'Donnell, 21 hours 21 min. 41 sec. Third, Amelia Earhart. In the light plane class, first was Mrs. Phoebe Orelle, in 25 hours 12 min. 42 sec. Second, Tina Randee, 31 hours 12 min. 30 sec. Third, Mrs. Keith Miller.

Arrival of AE Ohio Derby. First, M. A. Speer, 4 hours 22 min. 39 sec. Second, Lewis Love, 4 hours 43 min. 15 sec. Third, Barney Zanevsky, 5 hours 13 min. 1 sec.

Liberty Engine Bankers Trophy Race, 120 miles, 12 lap course. First, Maj. J. S. Owens, 143.07 m.p.h. Second, Benjamin Xmas, 133.72 m.p.h. Third, John Gil, 132.72 m.p.h. Mrs. Owens and her Filling got away to a brilliant start with two victories in the first two days.

THE FIRST EVENT on Tuesday's program was a race for women contestants only, ten times around a 3-mile course and using planes powered with engines of 500 cu. in. displacement or less. It marked the first appearance in this category of women pilots in closed-course racing. Mrs. Keith Miller, flying a Fleet biplane powered with a Kinner 325 engine, was the winner, with an elapsed time for the course of 30 min. 53.2 sec., and an average speed of 98.73 m.p.h. Second place went to Lady Mary Heath in a Great Lakes Trainer, tied with an American Cirrus engine, flying the course at an average speed of 96.17 m.p.h. Miss Blanche Noyes in another Great Lakes plane took third, averaging 85.12 m.p.h.

Mrs. Phoebe Fairgrave Orelle in a Warner-powered Monocoupe actually made the best time over the course, but was disqualified for having crossed a pylon. Her time was 26 min. 41.7 sec. and her average speed was 112.38 m.p.h. Mrs. Orelle filed a protest against the ruling, stating that she turned around and circled the pylons before carrying on with the race. This protest was substantiated by Mrs. Keith Miller and Lady Elvira. The Contest Committee of the N. A. A. has not yet ruled upon the protest. A distressing incident throughout was



A bird's eye view of the 1929 National Air Races held at Cleveland, Ohio.

the delay in getting rulings on protests and other controversial questions, or failed in securing any sort of official judgment on the protests. The top times and speeds on the closed-course runs at the first day of the meet were not yet released and ready for official release when the week of racing was over.

A fifth entrant in this race, Miss Amelia Earhart, who flew a third Great Lakes Trainer, was also disqualified for having covered another entrant on a turn.

Despite the limited experience of racing that most of them had been able to secure, the women showed an average of skill and intelligent handling, superior to that of the male competitors. A conspicuous feature of the race was the flying of Lady Heath, who was seldom overshooting the pylons by more than twenty yards during any part of a turn. She was completing each turn, from going to the pylons, the fourth of one lap in sweeping around it and passing it again straightened out for the next lap, in approximately 4 sec. time which only half a dozen or six of the male pilots were able to best consistently. Some of the hand experience of her competitors were using as much as 12 sec. for each turn, in itself enough to cost them 6 or 8 m.p.h. of average speed on a 5-mile triangular course. Lady Heath, who at some had had a very extended racing experience in landing events in England was banking about 60 deg and diving steadily around the turn for a total loss of this rate of some 75 ft. in each turn. Mrs. Orlin banked her Monocoupe only about 35 deg. and dived into each turn and dove out of it, a critical mistake to many of the pilots. Mrs. Miller's turns were notably good, although not up to the standard of her British rival.

The feature of the race was the rising ability displayed by Lady Heath who took the pylons in clean style. Lady Heath is of course not the last of the air racing genre, having been met in many of the light plane events on the other side of the Atlantic.

Due to the direction of the wind while the women were racing, the Navy pilots who were going aloft for their daily exercises above the beach of the crowd were forced to take off directly in front of the stands. In so doing they had a smaller chance of that which constituted a large part of the field near the home pylon and made it extremely dangerous for the women to make the turn, to say nothing of making the race visible to the spectators. Incidentally this was most disagreeable throughout the entire race week, getting worse toward the end. It is to be hoped that the time is not far distant when airport engineers will have solved this problem which is a dice menace to the participants in air races and an unfortunate annoyance to those who have paid money to be entertained. Incidentally, the same ground prize for spectators should have gone to those who put up signs, flying a grandstand periodically cut off from the field by opaque brown clouds, proclaiming:

"The National Air Race Association uses — to eliminate dust."

STILL FOLLOWING the finish of the women's race, S. A. Walks roared across the field in his Travel Air D-4000, powered with a J6 Whirlwind, to take first place in the Portland, Ore., to Cleveland derby open to planes powered with engines of 720-800 cubic displacement. Wolff's flying time for the one-hour flight was 14 hours, 44 min., 10 sec. Then men were badly contacted by an unfortunate series of days completing the course. Charles ("Speedy") Robinson, who was in second place at Mil-

waukee, was forced down in Stanbury, Ohio, a few miles from Cleveland, and surrendered his position to Les Rankin in a J6-powered Waco Tiger Wing. Rankin held his position to the finish and closed the race with a total time for the route of 15 hours, 30 min., 16 sec. Third place went to Spedder Hall, flying a Travel Air Speedwing fitted with a J6. Hall's elapsed time was 17 hours, 25 min., 30 sec. The rest of the field finished in the following order:

W. H. Tazewell, Jr., Travel Air BTD, J6 engine, elapsed time 17 hours 42 min., 22 sec.; Maj. G. H. Robinson, Waco Tiger Wing, Whirlwind engine, elapsed time 18 hours 14 min., 55 sec.; Dick Rankin, Waco, Whirlwind engine, elapsed time 18 hours 41 min., 28 sec.; W. B. Clark, Travel Air biplane, Whirlwind engine, elapsed time 23 hours 35 min., 26 sec.

One of the most interesting events of the day was a 30-mile race for planes with engines of not more than 100 cc. in displacement, a crossover of the light plane events started at Dayton in 1924. Three pilots were entered in this event, first place going to K. H. Hensley in a Heath Parsonal monoplane powered by a 27 hp. Henderson engine. His time was 47 min. 41.3 sec., and his average speed was 64.91 m.p.h. The other two entrants in this race were Freddy Land in a Heath low wing sport plane and Jack Miller in a Super Sport plane with a Henderson engine. The Heath Baby Model for some reason did not appear. The distinctive planes kept the crowd at great attention as they sped around the pylons but so many spectators headed for a local party.

Another event of a similar class was a light airplane speed and efficiency contest open to planes powered by engines of 275 cc. in displacement, or less. In this contest each of the five planes entered was required to carry at least 0.2 lb. of payload for every cubic inch displacement of its engine, the efficiency formula being as follows:

Pay Load X (Miles per Hour)³ equals Figure of Merit
(Gas Consumed)

Since all of the planes with low-powered engines protested that this formula, which was used in a number of contests during the week, gave too much weight to speed, theoretically it should equalize perfectly the effect of change of weight, other things being equal, and it is much wonder and more logical since most of the "efficiency formulas" that have been developed and tried in the past. Undoubtedly, however, it strongly favors a plane with high wing loading and high landing speed. Plans were awarded both for speed and efficiency. H. A. Speer, flying a Barling N8-J low wing monoplane, powered by a Lilland engine, taking first place in both divisions. His elapsed time over the 30 mile course was 38 min. and 34 sec. and his average speed 105.2 m.p.h. William Rose, in the Heath Baby biplane, took second place in the speed division with an average speed of 104.08 m.p.h. An Inland Sport monoplane, flown by J. W. Eggle, covered third position with an average of 100.62 m.p.h. Fourth place went to Lena Loe, taking a Davis V3 monoplane, whose average speed was 91.43 m.p.h. In the efficiency scoring second, third, and fourth places went to Dugle, Burns and Love, respectively. Barney Zimmerman, flying a Barling N8-J, was forced out of the race on his fourth lap with engine trouble.

The biplane, with its twenty-foot span, landing gear shed with fuselage ends which completely reduced its fabric, and a fully evolved Bristol Centaur engine driving

a 32-hp propeller, again startled the crowd as it had done at Los Angeles. The overall length of the machine is only about 43 inches, and the rear finest of the fuselage have about five inches of ground clearance.

Both Barlings were flying very low and turning easily. Up to the time of his forced withdrawal Zimmerman seemed to be getting around more quickly and smoothly than Speer, who was too low to clear the roadway into a bank with comfort without a preliminary climb. There was also keeping near the ground and climbing sharply into his turn, which was very wide at first but improved notably as the race progressed. Zimmerman's name to go around a pylon, the Lady Heath's, was a 4 sec.

Thursday also was featured by the arrival of the Miami-to-Cleveland derby fliers. This derby was divided into two classes, Class B being limited to planes with engines of more than 225 and not more than 510 cc. in displacement, and Class C for engines with a displacement ranging between 510 and 720 cc. in. George E. Huber, flying a Challenger powered Hispano Bee engine, won first prize money in the Class B field. Huber's elapsed time was 12 hours, 42 min. and 545 sec. Other prize winners in this class in their order of standing were C. D. Barling, Curtiss-powered General 12 hours, 59 min., 1075 sec. and J. Carroll Case, Challenger Commodore, 13 hours, 30 min., 355 sec.; Leslie H. Bowman, in a Kaiser Monocoupe, took first place in the Class C race, his elapsed time being 12 hours, 44 min., 235 sec. A Warner powered Avenger, Moth, flown by Robert Dale, took second place in this class, while third place went to Charles W. Meyers in a Curtiss powered Great Lakes Trainer, and fourth position

to Earl Rowland, who flew a Warner-powered Contest. Cal Charles A. Lindbergh, who had made his recent appearance at the meet on Monday completing one of Tuesday's major thrills by going aloft with two members of the Navy "High Hat" squadron, from San Diego. He led them in a remarkable exhibition of close formation flying, as well as group acrobatics. The Oakland "High Hat" truck included in the race was leading the V formation back and forth before the grand stand, and a fourth taking landing in front of the stands immediately after cutting his engine while finishing a lap. With Lieutenants F. O. Kovacic and Frank O'Brien, of the new High Hat, Oakland Lindbergh made the 15 min. flight a daily feature at the race thereafter. The planes used, of course, were Boeing fighters of the E2B model, the first to go into production with a Wasp engine. Tuesday's exhibition marked Lindbergh's first appearance as a member of a naval team. It is to be recalled that he had two previous Army days were instead attractions at the Los Angeles races last year.

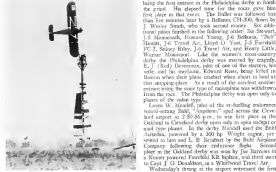
Another accomplishment, that is worthy of mention, on Tuesday's half was a series of seven consecutive outside loops hunk by Bert Blodde in a J-6 powered Waco Tiger Wing. Blodde's display was an official record for such things. Blodde was credited with having set a new world's record for outside loops, the previous record of six having been held by Lieut. A. Williams of the Navy.

WEDNESDAY'S AVIATION festivities included the arrival of the Philadelphia and Oakland derbies, as well as the arrival of the United States Navy's drizzle "Los Angeles," and drew probably the largest attendance of any day of the races. Between 85,000 and 100,000 persons were estimated to have visited the airport.

Eug Williams, at the controls of a 36-powered Eagle-Nord Balder, roared across the finish line in 2:35.48 p.m., being the first entrant in the Philadelphia derby to finish the grand. He elapsed time for the route gave him first place in that event. The Balder was followed less than five minutes later by a Bellanca CH-360, flown by "Wesley" Smith, who was second in this event. National pilots finished in the following order: Joe Stewart, J-5 Monocoupe, Howard Young, J-6 Bellanca, "Bob" Hewitt, J-6 Travel Air, Lloyd O. Yast, J-5 Fairchild PG-2, Sidney Riley, J-6 Travel Air, and Henry Little, Warner Monocoupe. Like the western monocothers of the Philadelphia derby was earned by tragedy. E. J. (Red) Devereaux, pilot of one of the starters, his wife and his mechanic, Edward Ross, being killed in Boston when their plane crashed when about to land at after stoppage-point. As a result of the accident another entrant using the same type of monocothers was withdrawn from the race. The Philadelphia derby was open only to planes of the monocothers.

Leslie W. Hendley, pilot of the re-buffing endurance record-setting Bell "Angelinas," sped across the Cleveland airport in 2:46.50 p.m. to win first prize in the Oakland to Cleveland derby open only to open cockpit or mud type planes. In the derby Hendley used the Bell Aerodrome, powered by a 300 hp. Wright engine, pre-owned to him and L. B. Rembert by the Bell Aeroplane Company following their previous flight. Second place in the Oakland derby was won by Joe Barling in a Kaiser powered Fairchild KR biplane, and third went to Capt. J. O. Donaldson, in a Whirlwind Travel Air.

Wednesday's throng at the airport witnessed the first public exhibition of an airplane being attached to an



A vintage plane awaiting the race pilot

simply in flight. Learnt A. W. Gordon of the Navy, flying a Vought fighter, accompanied the feat at an altitude of about 10,000 ft. just behind the gliderman, after two preceding attempts made over the airport had failed. With the propeller of the Vought idling and the Learnt A. M. Bolander, of the glider's crew, climbed down through the bottom of the netting and became a passenger in the airplane. After nearly 30 min. of being carried along as a kite on the Los Angeles airplane was dropped loose, dived for flying speed and clearance, and then creaked down to a safe landing before the astonished crowd. Lieutenant Bolander was introduced to the spectators as being the first man to be transferred from a glider to a heavier-than-air craft while both were in motion. After the transfer was made the guest airplane also descended and was recovered, until the following morning, at the combination mooring mast-pylon before the stands.

THE WEEKEND AFTERNOON PROGRAM also included the OX-5 race which had been run off on Sunday but had been declared no contest because some of the pilots flew around the 10-mile pylon instead of the 3-mile markers. This event, open to all type planes powered by Curtiss OX-5 engines and having two or more seats, included 8 types of a 5-mile course, and was won by Douglas Davis, in a Travel Air, who flew the course at an average speed of 104.90 m.p.h. and in an elapsed time of 32 min., 37 sec. W. E. Winkle, in a Duane-Wiggle "Bird," was second, with an average speed of 101.74 m.p.h. and a time of 34 min., 10 sec. F. B. Sprenger, in a Curtiss Robin, finished fourth with 98.32 m.p.h. There were eight entries in the event.

Douglas Davis' team was again conspicuously good, but the general average was much better than the first day, and few pilots were reporting over seven sec. for a turn. Eric Jackson, in a Curtiss Robin, was getting around with very little lost motion as long as another Robin managed to stay in front of him, but after entering up with it and proving it he began to overbook the marks a little more. Mr. Winkle, flying one of his own planes, made by far the most consistent turns, although not the fastest, throwing the plane very sharply over into a bank sometimes passing well beyond the vertical. The general tendency was to fly lower than during the first attempt to run the race.

Shortly before noon, and before the huge afternoon crowd had gathered at the airport, a transport was loaded with fuel and supplies from an army Douglas observation plane to the Boring mail plane "Shuttle" being flown across the nation by Capt. Ben Zakler under sponsorship of the Army, the Frontier Department, and the Boring company to test the feasibility of feeding, loading and discharging cargo in flight. The Shuttle at the time was enroute to New York on its next to last journey. Later the same day, after having returned from New York,

the Shuttle was forced down on the Cleveland airport when an oil can accidentally dropped from the re-berbering crew struck the lower plane's wing and broke an oil line.

Thursday's racing program at the airport began with a 60-minute race for women pilots only. Six entrants started, the contest calling for 12 laps around the five-mile course, and for planes equipped with engines having a price displacement ranging between \$10 and \$800 cost. This event furnished the few pilots with their second opportunity to display their skill at closed-course airplane racing—and the skill was displayed. Considering the equipment the race was flown at an exceedingly high average speed, and a number of the girls showed an amazing ability to control the plane with a minimum loss of time. Gladys O'Donnell, of the cockpit of her Waco J-5 taperwing, who took the lead right at the start, sped over the finish line, the winner, with an average speed of 127.6 m.p.h. Just behind Mrs. O'Donnell, and taking second prize, was Mrs. Louise Thaddeus, winner of the women's Los Angeles-Cleveland derby, who flew her J-5 powered Travel Air equipped with an N.A.C.A. wing at an average speed of 121.43 m.p.h. Miss O'Donnell's elapsed time for the 60 miles was 26 min., 0.8 sec., while that of Mrs. Thaddeus was 27 min., 25.5 sec. Mrs. Blanche Norris, in a Travel Air J-5, also having an N.A.C.A. wing, took third place with an average speed of 122.77 m.p.h. The quality of the flying in this event was better than in the ladies' first attempt. A shadow was cast over the women's competition by the news of the serious accident that Lady Heath had suffered on Thursday morning, for her sister, in a Curtiss biplane, was still being very anxious for several days thereafter.

THE NEXT RACING EVENT was a 10-lap grid around a 5-mile course and open to planes powered with engines having a power displacement of between 275 and 310 c.u.t. There were so many entries in this event, known officially as Event No. 9, that the race committee was compelled to settle the matter in two elimination heats and a final. In the initial heat, held Thursday, 11 planes competed, eight of them qualifying for the final. Those proving capable, in order of speed shown in the first heat, were: Vance L. Roberts, Warner Monoplane, 129.58 m.p.h.; R. T. Qualey, Warner Monoplane, 130.06 m.p.h.; C. E. Clark, Chevrolet low-wing Travel Air, 126.14 m.p.h.; Charles W. Meyers, Curtiss Great Lakes Trainer, 122 m.p.h.; W. G. Moore, Warner Monoplane, 116.72 m.p.h.; H. A. Latta, Warner Monoplane, 114.34 m.p.h.; Leslie Roemer, Warner Monoplane, 114.06 m.p.h.; and A. P. Krogsh, 112.22 m.p.h.

The crowd was then treated to another display of speed by planes of the elite class. Event No. 5 calling for ten laps around the five-mile course and open to planes of all types equipped with engines of 275 cc. in displacement or less armed with six en-

gines. E. B. Heath in his Baby Birdlet, spot around the course at an average speed of 100.17 m.p.h. to take first place. A Baring X-32, LeBlond powered and piloted by H. A. Speer, was second with an average speed of 105.03 m.p.h. and an elapsed time of 28 min., 10.2 sec. Third place went to Lewis Love in a Deere monoplane, also equipped with a LeBlond. Lewis' average speed was 103.6 m.p.h. and his elapsed time was 28 min., 24.4 sec.

ONE OF THE HIGHLIGHTS of the afternoon was the arrival of the Toronto-Cleveland defiance. The first of them, known as the Canadian Commercial Derby, had its entries and was won by Herbert St. Martin in a Waco Travel Air who flew the route in 2 hours 10 min., 24 sec. W. J. E. Johnston in a Buhl Arrowhead, J6 equipped, took second honors in 2 hours 13 min., 14 sec. J. G. Craig in a J-5 Buhl and T. F. Williams in a Times Argonaut finished third and fourth respectively. Only four of the six starters finished.

The second derby, known as the Canadian Club derby had four entries, all of whom used Graysbi Models and finished in the following order: First, K. E. Whyte, 3 hours 26 sec.; second H. C. Green, 3 hours 2 min., 49 sec.; third, G. M. Irvine, 3 hours 14 min., 36 sec.; and fourth, L. J. Trapp, 3 hours 19 min., 32 sec.

The third closed course event of the day was Event No. 15, a 10-lap 5-mile course race open to planes powered with engines of 720 c.u.t. displacement or less. After a beautiful exhibition of plane turning and pushing up speed in a dive from the straight-away, Eric Winkler, second his 90 Eaglehead Bledit up from the field with first place in his possession. Williams made an average speed of 134.58. Second place money went to E. V. L. Roberts who also gave the crowd a lesson in plane turning in his Warner powered Monoplane. His average speed was 138.84 m.p.h. R. T. Qualey, a similarly equipped Monoplane took third place money after receiving on Robert's tail with an average speed of 128.15 m.p.h. C. D. Royer in a Warner Consul just dipped into the money with an average speed of 126.41.

There was a total of eleven aspirants in this event which resulted in many exciting gap turns around the home pylon. As a matter of fact it was plane-turning ability that enabled Williams to bring home the bacon. During one of these thrilling postures H. S. Myhrum in a Simplex monoplane, all pointed up like a race horse, failed to come out of his pylon dive and bounced by wheels off the ground in a cloud of dust. For a second it looked as though Myhrum were headed for a crash, but instead he gained high enough to make a 90 deg. turn into the wind, and brought the Simplex down on one wheel and a wing. "Bounce" Taylor, the announcer, termed it "no race, no hit, no crash."

The closing exhibition of the day was a shooting show by three Canadian Royal Air Force pilots, Flight Lieutenants F. E. Beazley, E. A. Baring and G. R. Houston. Their planes were Armstrong-Whitaker Siskin fighters, all-steel single seater, powered by 425 hp. Jupiter engines. The show that they put on was without question the most thrilling of any team flying during the week. Very little time was spent in formation flying, in fact, only enough to permit the visiting two to give sufficient altitude. Then they broke loose and proceeded to show Cleveland and its visitors how they

shoot in Canada. Not a single mark was missed, and as a matter of fact a new act in American plane was demonstrated. This was called the "double loop." It starts with half an outside loop, and when the pilot is on his back at the bottom he cuts the gun and flaps across the field, after which he half rolls over. The thrilling part of the exhibition was when the two men crossed the crowd and variously rolled their wheels off the track lines. When there was hardly a breath left in the crowd, the two landed in formation on the far side of the field and stood up to the stands. By that time the spectators were almost back to normal and



Mayor Marshall (second from left) stands next to the B-18. Subsequent to the race.

they proceeded to give the visiting guests the best haul of the meet thus far. A wonderful exhibition, but one that was far too close for comfort and safety. It set a bad example.

When informed that the Solon planes are high altitude fighters and are not at their best near the ground, one prize any entry plane that might run afloat of the Canadian boys that visited Cleveland.

Although the Mason M. Patrick trophy race and Mary's pursuit race, both regarded as actual classics in the reputation bracket of the Stevens race, place Friday afternoon, probably the major thrill of the day was served the crowd by a lone pilot in a three-to-five-gal. monoplane plane several thousand feet above the field. This pilot, one Charles "Speed" Holmes, who has broken into print on a number of previous occasions by his unusual aerial average speed, proceeded to prove to the 20,000 or more spectators that a Ford all-metal monoplane is not nearly so stout and disguised as it looks. As a surprise feature of the afternoon's entertainment a had been assumed that Holmes would take the Ford up on a starting expedition. Since Ford planes are known to be designed for more practical purposes, it really was a surprise when Holmes began to make the monstrous plane an expert. Altogether the Ford, powered by three Wasp engines, cleared a total of five perfect laps, several times the length of the race, in one, a still at two, and best of all, it flew at about a constant speed again, down, that flight being being while on top of a loop. The crowd took so well to this amazing demonstration that Holmes repeated the performance the following day. Friday's performance was



Planes of the Navy "High Bird" squadron attended to the "Hammonds" (back up the "Herring")

not the first time a Ford had been loosed, nor the first time Hoffman had loosed one, but it is safe to say there were not more than a score of persons among the thousands on the ground who had ever seen one started before.

Another interesting solo exhibition started Friday afternoon was the flying of Lt. Harry Johnson, of Selfridge Field, Mich., in a Curtiss P-1, equipped with a Pratt & Whitney 12-12 engine. A third was a demonstration of prolonged inverted flight by Lt. "Tex" Rogers, of the Marine Corps, in a Marine Corps Curtiss Hawk, equipped with an especially built carburetor and oiling system. In Johnson's plane the radiator is just normal size and filled with Pratt & Whitney liquid designed to keep the engine operating at a temperature of 300 deg. F.

Thirteen pilots of the Third Attack Group from Fort Crockett, Tex., participated in the Navy Mid-Park trophy race, which is limited to members of that group. The race consisted of 8 laps of the 30-mile course. The trophy in this race was donated by the Hon. F. Theodore Dawson, assistant secretary of war as a personal tribute to the former chief of the United States Army Air Corps. It was accepted last but the first time at Los Angeles last year.

Popular spectators notwithstanding, the race was only won by Lt. Col. Roy M. Palmer, whose plane was the much classed number 12. Lieutenant Palmer's average speed was given out as 193.2 mph, which incidentally, is just seven tenths of a mile faster than the time set by Lt. G. R. Adelson, who was the event last year. The best time took the lead shortly after his dive for the opening pylons and he kept it consistently. He averaged a speed of 152.72 mph on his fifth lap and reached his highest speed on the seventh with an average of 184 mph. Second place in this race went to Lt. Col. P. K. Kowalski, whose average speed was 136.81 mph. Lt. Col. M. Ziegler flew the eighth at an average speed of 136.14 mph to take third position.

The next racing event of the afternoon was a relay race for male contestants, each plane being manned by four pilots, the pilot being relieved after three laps around the five-mile course. A final lap was flown by pilot Douglas Davis, C. D. Clark, George Galt and J. D. Goehry, won the event in an elapsed time of 39 min. 12.6 sec.

SCHEDULED THE EXACT race the semi-final contest of Event No. 3 was flown. Out of eight starters Gene C. Clumberlin, in a plane with six new wings, graded out the race with an average speed of 114.8 mph. Chester Barila elapsed time was 26 min. 07.9 sec. Porter Cramer, at the controls of a Warner-powered Cessna averaged 112 m.p.h. to come in second.

Final the standpoint of expected speed the next event, the Navy Futrell Place race, was a relay race, sponsored by Hawkeye, due to the number of persons entered and the unusually large number that fanned themselves jockeying for advantageous positions on the pilot's turn, the race proved highly exciting to those in the stands. It might be explained here, as it was explained to the spectators, that the winging pilots, coming from San Diego, flew a new and somewhat better route to the Air Show, and, of course, wanted to return to their base following the Cleveland class. This fact, no doubt, accounts for the fast start, without reception,



The Navy mid-air shipy posing over the judges stand

the 14 pilots flying the 300 mile course seemed to hold their charges in check. Lt. Col. H. E. Arnold, flying plane No. 19, managed to get his nose on the arch covered trophy by covering the 100 miles in 43 min. 04.1 sec., averaging 129.58 mph. It will be recalled that the winner of the same race at Los Angeles last year averaged 148 mph. Approximately 12 seconds behind Arnold, Lt. Col. C. Ironmonger started across the finish line to take second place. Lieutenant Ironmonger's elapsed time was 43 min. 36.9 sec. and his average speed, 126.97 mph. Just to prove that one cannot keep a good Navy man down, Lt. Col. K. Luckfield, still handicapped from a minor cockpit experiment just before the San Diego crowd reached Cleveland, posted his flying F2H into third place, with an average speed of 120.5 mph.

Saturday was termed Engineers Day and all teststers were dedicated to Thomas A. Edison. Mr. Edison and Henry Ford and Harry S. Pratt were to have been the guests of honor, but due to illness Mr. Edison and Mr. Ford were unable to be present. Mr. Pratt, however, was among those present and the crowd showed its appreciation when he was presented from the engineering platform. Other celebrities who attended included Assistant Secretary W. P. MacCorkle, Jr., Assistant Secretary F. Theodore Dawson, and Senator Elmer Benson.

Several hostilities opened with a four-minute Army Percent plane race of 10 laps over a 10 mile course, for the Joan B. Mitchell Trophy presented by Gen. Wm. Mitchell in honor of his brother killed in France. This event has been completed for some 100 years and is the goal of every Air Corps pilot. In last an Army pilot won the Joan B. Mitchell Trophy as badly as the old commercial races wanted their names on the famous race of "Casey" Jones.

These were engine boys with D-12 Curtiss Hawks that took part, and they were all from the east of the Air Corps' much pursued pilots, Selfridge Field, Mich. The event was started by the entire group taking off in "V" formations of three planes each, two Vs getting off at the same time. When all were off and well clear of the field the leader, Maj. Ralph Rogers, winged his wings and dived for the home pylon. The second plane followed suit and so on until all were roaring down over the heads of the crowd. So many competitors in the race that by the time the last plane had set upon its way Major Rogers could be spotted banking around the third pylon.

From then on it was a great and thrilling game of "pyles, pyles, who's got the pylon?" At no time were all of the planes out of sight and on several occasions four and five planes roared in the home pylon in a bunch. To anticipate times for presentation, after having had a hundred miles in which to spread out the group of three ships crossed the finish line within a total time of 06 sec., spaced about 25 yards apart, and five finished within a total of five sec. Considering the amazing proficiency of the members of the group in formation flying, their work on the pylons was very spotty, and there was a remarkable lack of agreement about the best way of making the diving turn. The planes were exceedingly good, albeit distinctly slow, and the time for completing the swing around the home pylon ranged from 24 to 11 sec. Angles of bank were from 60 degrees to beyond the vertical. It would be impossible to pick out any one pilot as doing the best work, but the turns were by Lt. Col. A. L. Moore, Morgan, Williamson, Pringle, and Schoenfeld were particularly impressive. Lt. Richard E. Cobb gave the crowd instant heart failure on each lap by doing roll-over-rolls and coming out at least five to thirty feet above the ground. His time on turns was shorter than any of his competitors could show, but the plane was pulled around so sharply that there must have been serious loss of speed and there certainly was a very substantial loss of altitude which had to be regained by climbing in high straightaways. As he finished, Lt. Cobb called his machine 90 degrees and, as he was about to start a second round, he was in a vertical plane. He had narrowly escaped a crash near the end of the race, when a commercial plane took off across his course just as he was rounding a pylon. The Army pilots almost without exception avoided one another, crossing the civilians, that of starting the last two laps. Most of these flew from 100 to 200 ft. high and began to roll their planes well after a couple of hundred yards short of the pylon.

The winner of the military percentage race was 2nd Lt. P. B. Warrall. His time was 36 min. 0.8 and his average speed 132.123 m.p.h. The losers for second place were split by 2nd Lt. A. L. Moore and 2nd Lt. K. E. Rogers. Both piloted their Hawks over the route in 36 min. 87 sec. at an average speed of 130.04 m.p.h. Third place was won by 2nd Lt. E. K. Warrington who hung up an elapsed time of 36 min. 11.9 sec. and an average speed of 131.816.

A COMMENTARY of the times set up by the first lap in the Army men gave one a good idea of how evenly the Hawks were matched. An item of interest, though, in connection with the speeds shown is that although the winner's speed was the highest of the meet so

regards seven races, it was the slowest of any Mitchell Trophy race in the last six years.

While the Army boys were adding round the pylons the competitors in the Second the 8th of Ohio Derby set their wheels on the "Post" and started up to the judges' stand. Four of the boys had hopped off on Friday and three of them finished the race. The winner was J. O. Donaldson, who piloted his Jo Jo Aero Travel Aero over the route in an elapsed time of 4 hours 48 min. 11 sec. Next in line was W. J. Harrow, who set up a time of 4 hours 53 min. 47 sec. in his War powered Fairchild craft monoplane. And the last lap "in the money" was Larry Moore, whose time, however, made the race in his Jo Jo Bird in 5 hours 3 min. 9 sec.

THEN THE MARINES proceeded to "test the situation well in hand" by staging a six plane, 8 lap, 10 mile course race. The pilots used the standard Marine pattern first time through the course. Like the Army they also took off in formation, then strung out in line and went for the first pylon in a hurry. As there were only six entries there were very few cases of such hand-to-hand as had made the Mitchell Race somewhat. Most of the time they were in "follow the leader" style. Lt. Harry Rogers earned off first (time by average) the route in 30 min. 46.4 sec. at 142.28 m.p.h. Second place was won by 2nd Lt. W. C. Brice in 30 min. 59.4 sec. with an average speed of 141.88 m.p.h. and third place went to Sgt. B. F. Becker with a time of 31 min. 12 sec. and an average speed of 140.83 m.p.h.

The Marines showed much more consistency in their racing policy than either the Army or the Navy. One might almost have suspected the men of being released in an advance, as uniform was the style. Starting the race somewhat later and more sharply than most of the Army pilots, they came up to the vertical or within ten degrees of it and then swung around without any preliminary dash and with a loss of about 50 feet of altitude at each pylon. The three prize-winners were particularly close to type in their racing tactics, and they were particularly good, their turning time, as previously mentioned, ranging from 33 to 40 sec. The time of the Army officers except Lt. Col. Cobb and Moore.

While the Marine men were in the air, one of their brother officers, Lt. "Doc" Rogers, was showing the crowd how one flies on one's back for an indefinite length of time. Lieutenant Tex had a Marine Hawk especially equipped with a speed meter and carburetor systems which enabled him to carry on with full engine when on his back. More eyes were on Lieutenant Tex than on the racers. In fact the race was over before the crowd had ceased to get an eyeful of the "upside-down" pilot.

Because the crowd had been able to move or just follow the progress of the races past on doing the first part of the afternoon, the contest continued proceeded to hold an event known as an Australian parent race for men. The race was a 15-lap handicap after over a 5-mile course and the winner was a woman, who had been passed by another, it was to drop out. All entries had been flown for landing, start, and were lined up in order and flagged off at intervals. The handicaps were as varied that some planes had cranked the course several times before others were even started. And as the spectators, or at least those without

previous experience in figuring out today's bareback races, went on a tour trying to decide who was ahead and who was behind. On several occasions the "leader" would come roaring by, and then when he came around again he would glide down to land, and the crowd would gasp again. Finally the judges announced that A. P. Knapik is a Gypsy Moth hawk. His actual time was 39 min 29.9 sec, his handicap time was 30 min 30.8 sec, and his average speed 124.22 m.p.h. Second place went to J. Carroll Case in a Firebird KB with a Challenger engine. His actual time was 39 min 33.6 sec and his handicap 51 min 31.1 sec, and his average speed 112.8 m.p.h. Third place was won by the fastest pilot in the race an OX Eaglehawk flown by R. C. Waininger. His actual time was 51 min 32.2 sec. And his average speed was 97.32 m.p.h. The Eaglehawk, with the only win-tie engine in the race, had no handicap and was first in the air. In fact he was allowed such a head start that he threatened to establish a solo endurance record before the second entry was flagged off. Eleven engines started the race, but it was not long before they began to come down one by one. There was no spectacular turning, but those of Eric Williams in an Eaglehawk Buller and of Knapik and Waininger were very good.

BEFORE THE TOWN had had a chance to get back to normal after trying to figure the race's Australian pursuit race, several members of the fair sex were up and started off as a women's Australian Pursuit race. They were scheduled to fly 12 laps over a 5-mile course. The winner was Gladys O'Donnell who flew the route like a veteran. In fact, Miss O'Donnell took the home-gym on several occasions in less time than was sought for an outer competitor at the event, man or woman. She brought her Waco type wing around very smoothly with about 75 degrees bank and very close indeed to the pylons, being about 30 ft. of altitude steadily during the turn. She made one turn in 2.2 sec. and a number in under 3 sec. The next best time was of Mrs. Min Thidley, who used a Travel Air and whose turning time was a little longer. She probably lost somewhat more speed in turning, as she climbed in approaching the pylon. Freeman Hurrell, flying a Moth, was also very good. Miss O'Donnell gained on her field so rapidly that there was no reply to the spectators about her having been the victor.

The main feature of the Sunday events was a speed

display of flying by Lt. Col. Alfred J. Williams, of the Army, and the high spot in this brilliant exhibition was his descent at the airport while flying the plane in the inverted position. Lieutenant Williams followed the general outline of the regular race course and made his turn with about a 30-degree bank while flying upside down. One of the turns was around the home pylon and it was smoother, done in the pylon, and generally accomplished with less loss of time and distance than 75 per cent of the world's turns around the same mark in actual competition, in spite of the unusual attitude of the plane and the fact that the pilot was on the outside of the bank.

Williams did not lose altitude during this inverted "race" around the course and in circling the crowd found him to be down slightly, coming out in an inverted position which developed into a vertical climb and an over into the normal horizontal position. His lap of tracks included, also, a beautiful and indispensible outside loop, a number of smooth, slow rolls, and an "upside-down" landing. Eventually he came in to land still flying on his back, then rolled over onto his "feet" within 300 ft. of the ground. He used a Curtiss Hawk. A member of the "outside loop" made during the week, incidentally, were quite sketchy, and ended up with part of an inverted spiral and part of a roll.

Not long after Williams' flight, Lt. Col. James H. Doolittle, of the Army, took the air, fresh from his mission into the Caterpillar Club. Just previous to Williams' flight, Doolittle had taken off in a modified Hawk which the Army had been using in Wright Field for experimental purposes. It was Provisional-ordered. He went about 5 miles west of the airport to practice a bit. While in a long straight power dive the wings weakened off and Doolittle had to jump. The accident was not seen from the airport, and the first knowledge of it there came when Doolittle entered the Army office and asked for another plane.

IN spite of that hair-raising experience and of an engine which was not fitted for upside-down work, Doolittle maneuvered well and executed some inverted flying, including a bare while flying upside down. The engine spluttered and lost power disconcertingly each time an inverted maneuver was tried.

Another amazing show of a somewhat milder nature, but nevertheless very impressive, was given by Douglas Davis on the Travel Air modified monoplane. Davis of

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the technically-minded spectators were a little surprised about the display because of the very flat angle at which the losing wings attached to the wing, in accordance to usual racing monoplane practice, but Davis did a sitting of loops and sharp rolls, both horizontally and upward, without a sign of trouble of any sort. The machine had such an enormous reserve of power that it seemed to travel in any attitude or direction, including upside-down flight and flights vertically upward, quite normally and under full control.

T. A. Wells, of Wichita, won the race for open cockpit planes powered with engines of not more than 800 cc in piston displacement. He flew the 150-mile in a Travel Air D-4000 (Whirlwind Nine) in the



Retaining the world club ability at the 1000 National Air Race.

elapsed time of 30 min 32.6 sec—a speed of 142.37 m.p.h. It is 30 miles, flying the 30 miles low wing monoplane with a Whirlwind Seven, was a close second, making the course in 30 min 24.7 sec with an average speed of 155.56 m.p.h. Sydney Hall flew a Travel Air Speedwing (Whirlwind Nine) to third position with an elapsed time of 32 min 6.8 sec and a speed of 140.12 m.p.h. "Speed" Holman made the fastest time on this race but was disqualified for leaving inside one pylon.

The other planes had their taxing time. Engines were required to be of 800 cc in piston displacement or less. R. W. Carroll, flying a Lockheed Vega with a Wipac wing that event with an elapsed time of 35 min 38.5 sec—and an average speed of 122.7 m.p.h. Close behind him came Renee Turner, in another Vega with a Wipac engine, his speed 150.15 m.p.h. J. Wesley Smith finished third in a Bellanca with a Whirlwind Nine at a speed of 135.6 m.p.h.

The fastest race in this event was made by British in the Bellanca with the winner, Clewett, making an second best in that respect. The Bellanca was being thought around the pylon in less than 4.5 sec, which would have been rated as good time for a small open-cockpit machine. Both Bellancas piloted by Smith and by George Haldemans, flew very low, Smith in particular travelling at about 40 ft. on the strathway and circling slightly on the turn, and both banked within about 10 deg. of the vertical. The Lockheed, especially Cartwells, were flown considerably higher and not banked as sharply.

Excellent piloting, especially at the turns, added to the firmness of his Eaglehawk Bellen crashed Eric

Williams to turn in the best speed in the Aviator Town and Country Club Trophy, speed and efficiency event. His elapsed time was 26 min 45 sec, giving him 130.73 m.p.h. for the 125 miles. The plane was powered with a Whirlwind Seven. Right behind him came a 35-mph (Smith) and a 35-mph (Wright). The lower was piloted by Verne L. Holman for a speed of 128.43 m.p.h. The second, flown by R. T. Umbley, was almost a minute behind, covering the course at the rate of 127.37 m.p.h. Holman did particularly well on his turn.

The highest score for efficiency in the Town and Country event was given unflinchingly to George W. Haldemans, flying a Bellanca, with a total of 253.78. Williams was second place with 192.65 points, and C. F. Boyer was third in a Cessna with 144.33 points. These figures are subject to revision. Haines were required to carry 0.5 pay load per cubic inch of engine displacement.

As already indicated, the best maneuvering around the pylons was that of Eric Williams, who had been consistently good in all his races throughout the week. He started with about 70 deg. of bank, rising very low and clearing over the turn, but later increasing his bank to about 85 deg. near the end of the race. Clewett's Chandler also flew wider and took the turns more gently, banking only about 60 deg. and also flying quite low. Staying close to the ground, indeed, was characteristic of most of the pilots of such monoplanes in most of the event.

Then Rache flew a Gypsy Moth to victory in the second best of the 60-mile Australian pursuit race for women. Her elapsed net time was 36 min. 6 sec. and her average speed was 99.72 m.p.h. Mrs. Thidley in a Lockheed (Wright 35C) was second with an elapsed time of 26 min 18.8 sec, and a net time (which includes the handicap allowance) of 35 min. 6 sec. Her speed was 136.68 m.p.h. Mrs. Keith Miller was third in a Pica (Kinner), and Gladys O'Donnell was last in a Waco Tagawing (Whirlwind). The flying of the women in this race was characterized by the confidence of their take-off, major losses, and the point again going to Miss O'Donnell, who was almost but not quite as good as on the previous day.

At 3:55 P. M. Henry J. Brown, of Cleveland, arrived from Los Angeles in the non-stop race, bringing his Lockheed Air Express (Whip) over the course in 13 hours 13 min 7 sec at the rate of 184.4 m.p.h. He won this event against Lee Schenck, the only other pilot to complete the event within the rules. Racers Turner arrived after the daily time limit had expired.

THE FIRST RESULTS of the Detroit News 30-mile transport and efficiency race awarded R. G. Lockwood, flying a Firebird 71 (Wipac) with first place. His elapsed time was 34 min 31.2 sec and average speed 127.37 m.p.h. George W. Haldemans was second in a Bellanca (Whirlwind Nine) with an elapsed time of 36 min 39.2 sec and speed of 119.07 m.p.h. J. Wesley Smith was third in a Bellanca (Wright 35C) with an elapsed time of 39 min 29.3 sec and a speed of 113.74 m.p.h. It was later found that Lockwood had been disqualified for taking too long a take-off run.

The unofficial efficiency results for this event gave Chester Chamberlain first place. He flew a Cessna, winning 579.94 points. Lockwood was second with 502.08 points and Haldemans third with 342.35 points.



Here's "Mystery" and his record book on the Travel Air "Mystery" plane.

These figures are subject to revision. Each plane was required to carry a minimum payload of 1,000 lb and get off within 900 ft at a speed of not less than 80 m.p.h. With a maximum take-off distance, neither than on "static" tests, the amount of pay load that could be carried obviously depended largely on the strength of the wind.

During the afternoon the First Pursuit Group, a Marine Corps group and the two of High Hats led by Colonel Landreth put on their usual exhibitions in flight. With a formation of six planes also did their stuff. The features of the First Pursuit Group display was a formation of planes arranged to spell "A C" for Air Corps and "A" for Army. The latter alphabetical experiment had been attempted once before during the war. While very good, it has been by no means perfect. On Sunday it was greatly improved. Of fourteen planes, only one was perceptibly out of position as they approached the stand, and that out but a few feet. A brilliant exhibition.

Arthur J. Davis of Lansing, Mich., took first place in the second series of the 75-mi. Amphibia pursuit race for men. He flew a Waco Tiger Wing (Whirlwind) near the Edison place as in elapsed time of 46 min. 25.1 sec. S. J. Whitman, of Ford da Lee, Mich., finished second in 49 min. 27.4 sec., while George Hildebrand was third in 50 min. 27.4 sec. with a time of 51 min. 36.8 sec. The speed of the planes was not compared, since it was a handicap event.

Aeronaut was celebrated on Monday, September 2, in recognition of the lighter-than-air movement, and as guest of honor the most enterprising H. Hugo Leichter, commander of the flying ship Zeppelin, Paul W. Leichter, president of the Goodyear Tire and Rubber Company and of the Goodyear Aircraft Corporation, and others prominent in the aviation world.

Dr. Leichter was fired by the city of Cleveland in addition to being present at the meet. He later flew away from the airport to Akron in company with Goodyear officials in a representative of his firm's aviation and air transportation. The Goodyear DeLorean Dr. Leichter was called in front of the judges stand.

On account of the number of men which had been postponed from previous days because of lack of time, the events for Labor Day were started at 8:05 a.m. instead of at noon as originally planned. Difficulty was experienced with the work of handicapping the contestants and delays ensued, making it necessary later in the afternoon to arrange for an extension of the races through one more day.

The feature of the day's racing was the free-for-all in which seven planes, including one representative each from the Army and Navy, were flown. Out of this race came the sensational victory of the so-called Travel Air Mystery plane, the wire-branded low wing monoplane powered with a supercharged Whirlwind. Niss and fitted with an N.A.C.A. cowling and a very completely streamlined landing gear. The wing, covered with aluminum and highly polished, and with the main nose of the wing cowl and the bright red fuselage and tail surfaces the machine made a very striking appearance.

An average speed of 194.9 m.p.h. for the 30-mi. course was turned in by the light glider. It was achieved in spite of the accidental cutting of a gyron, which interrupted turning back toward the start. The race was the highest speed ever achieved in the United States

by a commercial plane, and far exceeded the best times made by any machine, either commercial or military, throughout the recent races. It is also, the first time in the history of American air racing since a commercial plane has defeated both Army and Navy contestants in a free-for-all.

Douglas Davis of Atlanta, Ga., was the pilot. His elapsed time was 34 min. 05.9 sec. On his fastest lap he was clocked at 208.69 m.p.h. Second place went to Capt. R. G. Brenne of the Army, flying a Curtiss P-51A with Wasp engine, also with the N.A.C.A. or water-cooling. His elapsed time was 34 min. 42.4 sec. and speed 186.64 m.p.h. Rosco Turner finished third in a Wasp powered Vega, in the elapsed time of 36 min. 48.7 sec.—a speed of 163.64 m.p.h. Lt. Conde J. J. Clark, flying a Navy Curtiss Hawk with Pratt & Whitney D-17 engine, was fourth.

The playing was watched with special attention in the free-for-all, since it was to be presumed that the most skilled and experienced of racing pilots available would have received that assignment. In the first three laps Lieutenant Brenne had passed three competitors and Davis had passed two, and it became apparent, as indeed it had been considerably earlier, that the race lay between these two. Both of them flew comparatively high by the standard of previous races, varying between 200 and 330 ft. Their times were a little superior to those of any of their competitors, although the general standard was very high, and Lieutenant Clark's maneuvering of the Navy machine and that of the Stinson monoplane were also notably good. The time around the turn was running from 3.5 to 4 sec. for the two fastest machines, but the high wing loading of these planes of course made it hardly possible to throw them around so quickly as one would like to see. For example, Miss O'Donnell's Waco. The competing pilot, providing, also of the proper angle of bank was about 75 degrees, with the Stinson loading a little more steeply than that, and Roscoe Turner's Lockheed a little less. Both of these pilots tended to dash directly as their names, while Lieutenant Clark's direct about 60° on each one and Douglas Davis went around without appreciable change of attitude.

ANOTHER FREE-FOR-ALL of Monday's racing was the experimental one (or various in which four machines participated). It was limited to machines having a total engine displacement between 310 and 800 cc. at 10 S. Miles was flown in a Stinson high wing monoplane with Warner Scarb's engine, in an elapsed time of 10 min. 12.2 sec. and an average speed of 147.01 m.p.h. second the 20-mile course.

C. E. Clark in a Travel Air low wing monoplane with 6 cyl. Chevrolet engine came second with an elapsed time of 13 min. 13.8 sec. and an average speed of 112.28 m.p.h. In third place was C. J. Redburn in a Beach Baby Biplane powered with a Bristol Cherub. His time was 14 min. 16.8 sec. and his average speed 103.4 m.p.h. The fourth place, an N. B. 3 with Armstrong Siddeley engine, was flown by A. A. Spens.

The definition of "experimental" was a little uncertain, and it fell to the machines that participated in the race were entitled to carry in that class the same privilege could have been claimed for several others, both civil and military, of even higher speed.

Walter Davis and William S. Black, flying tri-engine Bessie, walked away with the multi-engine



Change-up of airplane wing attachment

race, detouring M. G. Galt in a Fokker by a wide margin. Waterman completed the course in 39 min. 4.6 sec. at an average speed of 136.6 m.p.h. Brock was close behind with an average speed of 134.46 m.p.h. Galt took 22 min. 20.4 sec. and was clocked at 123 m.p.h. Gladys O'Donnell won the Cleveland-Pittsburgh race for women, flying the 100 m.p.h. in 46 min. 5 sec. Louise McPhetridge Thelen finished second with a time of 50 min. 4 sec. and Ruth Nichols, the third contestant, arrived about 25 min. later.

WHILE THE RACES WERE GOING ON of considerable interest to both the flying spectators and the members of the industry, there were many non-racing events that will long be remembered. Perhaps the major ones were the stunt exhibitions, because of their spectacular nature and their appeal. Every day Colonel Landreth led his two High Hat friends into the air in Navy pattern planes and put on a show that included winged formations, spiraled cone loops, inverted flying, and other maneuvers not required by the Department of Commerce. The Army, Navy, and Marine Corps did their own every day as a means of showing top powers when the pilots could do with the artificial (or natural) show money.

Although all were very brilliant, and special scenes had been made of some of the special exhibitions given by the Pursuit Group, the first award for proficiency in taking up and maintaining formations went to the Navy's Fighting Squadron No. 1, "The High Hats." On several days toward the end of the week one division of nine planes maneuvered with the members of each section of three took together by ribbons at wing tips, a stunt done famous since 1919 by Lieutenants Doolittle and Brockway and the late Lieutenant Pearson and Captain Skid of the Army Air Corps. The Navy's Navy fighters not only took off and maintained their formation and turned in formation without breaking their 90-ft. ribbons, but actually changed from Vee to echelon and back again. On one occasion, when conditions were right, the whole division landed without

having broken a single one of the six connecting ties.

The warner's most spectacular contribution was a dual stunt-race, two sections of those performing the maneuver side by side and only a couple of hundred feet apart, the members of each section closing with other around and around a string of looms. Civilian stunt teams representing the Waco, Laird, and Travel Air Companies did their duty down also. Many of the stunters, both military and civilian, stayed so close to grandstands at times. A beginner might guess with stunts performed over each head, but if anything should go wrong it would be less tragic for one pilot and one airplane to suffer, especially as the parachute gives the pilot a chance, than to involve fifty or sixty innocent bystanders. As a matter of fact there were several cases when the stuntmen pulled up with almost nothing to spare.

Each day the program was opened with dead cat landing contests, and these events, held directly in front of the stands, met with the hearty approval of the crowd.

Each day also saw various problems stepping off into space pulling up cards, and endeavoring to land in a circle marked out on the field. Sometimes they missed the field altogether, and that added to the thrill of the affair.

As another frequent side attraction on the program four Goodyear biplanes flown around the field, in formation and otherwise. The Goodyear was taken aloft several times by its instructor, Sergeant A. C. Galt. These flights attracted considerable attention, as the Cleveland appearances were the first public demonstrations of the airplane in this country.

AVIATION MEN know was harassed by various glider flights. It had originally been planned to make glider flights a part of each day's program, but due to the failure of the program and the delay in getting some of the events run off the gliders were continually shelved, and, as a result did not get the attention they merited. Incidentally there is much room for improvement in regard to the control of the glider flights. As yet. While the rules state that all entries must be in by six o'clock on the previous evening, there were many cases of last-minute entries getting into a race. In fact sometimes even the officials themselves did not know all of the entries, weak less the type of machines they were flying, and in one instance the association's president a public appeal for someone in the audience to come down and tell him who was who around the course.

By actual count made one morning of an average day during the week there were a total of 417 planes of various types on the ground inside the airport fence. Although this total includes the 96 military craft, it does not take in the planes in the air over the field at the time, or the one Goodyear, three or four craft parked just outside the airport boundaries, not 8 or 10 gliders that contributed to the overabundance during the week. Classified by type the count revealed that 169 of the total were open cockpit biplanes, 120 were cabin monoplanes, 17 were open monoplanes, and 6 were cabin biplanes, making a total of 312 planes of the strictly high racing type. To be added to this was one low wing, four of which were Army or Navy owned and low wing, several. Out of the grand total, only 15 of the planes were of the tri-engine type. Twelve, including the seven Keystone bombers from Langley Field, were two engines.

THE S. A. E. DISCUSSION OF Aircraft

By EDWARD P. WARNER

IN ACCORDANCE with the practice at all of its sessions for the last two years, the S. A. E. concentrated its work on a discussion of the Society's standardization work in the aeronautical field. The only fixed element on the program was a report by W. M. Johnson of Nela Park upon the studies that have been undertaken by the aircraft lighting subcommittee of the S. A. E. Standards Committee. Opening his remarks with an assurance that there were hardly more air-headed aeronautical planes equipped with landing lights in the United States, Mr. Johnson explained that the aircraft lighting committee had undertaken to direct a co-operative research in which a number of aircraft operators and manufacturers of equipment had volunteered to participate. Operators who are taking part said they had planned to render reports upon their experience with the test equipment prepared under the direction of the committee at the Patuxent, National Air Transport, Boeing, and Stout Lines, and the Army Air Corps.

While the tests have not yet been carried far enough for a detailed report, the experience that is to be used was described. Special lamps have been made up to provide for any system of the inclination of the beam and for simple variation, by a change of the lens in front of the light, of the angle of the beam both in the horizontal and vertical plane. In the test by the air line operators, the planes will be equipped with one fixed light of the ordinary type and with one of which the direction can be controlled by the pilot. The beam being aimed anywhere between the horizontal and the vertical. The report will be varied from two degrees to 15 degrees above the horizontal to forty degrees, a well-defined one. Reports will be rendered as results are obtained and it is hoped that actual indication of the power and the other characteristics of landing lights at figures that will be satisfactory to all pilots will become possible through the experience gained in the test.

L. C. Porter discussed the report at some length, with special reference to the work of an aeronautical lighting committee of the Illuminating Engineering Society, of which F. C. Hargraves of the Department of Commerce is the chairman. The I. E. S. committee is at work especially on the theoretical aspects of the problem of illumination, both for the airplane and for ground lighting. Among the other topics that it is studying are the standardization of color of lighting for particular purposes, the determination of the best period of flash for emergency lights, the relative efficiency of concentrated and diffused lights for parking lots, a selection of the best

coloring and type for boundary lights so that they may readily be distinguished from highway or city lighting, and the ideal grouping of flood lights around a field. Mr. Porter spoke especially of the desirability of working for international standardization on these points. The chairman also emphasized the international aspects of the work, with special reference to the importance of securing agreement with Canada which will facilitate international operation across our northern boundary. The Canadian Government, in turn, naturally wishes to keep its practices in harmony with those of other portions of the British Empire and the quest for a general international standard becomes necessary.

F. G. Della urged the importance of immediate attention to the standardization of airport lighting, with special reference to the success of indicating the extent of the runways and the direction of the wind, and to the provision of means for signaling to the pilots in flight over the airport without having to use radio.

Anticipating International Movement

Across the general standardization topics taken up for consideration, the first was the mounting of instruments. A communication from the chief engineer of the Consolidated Instrument Company suggested something in all cases so that the instrument positions would all lie in the same direction for normal flight conditions and proposed that the figures on the dials should be arranged regularly as in a wheel chart, so that the instrument could be scanned in any angle for mounting without turning all of the figures upside down or making them appear unaccounted. Mr. Posey of the Pioneer Instrument Company expressed the belief that instrument manufacturers would be willing to put the figures anywhere on the instrument, and to make them read right side up with the instrument at any angle desired at very little additional cost to the manufacturer of an airplane in comparative production. Wesley L. Smith of National Air Transport who had declined at a similar meeting a year ago to express any definite judgment on the best way for instrument dial was prepared on this occasion to express his conclusions in favor of larger dials. In spite of the recent economies looked by the Army and Navy for material decrease in size as a means to more compact mounting, Mr. Smith felt that the strain on the pilot from peering at dials less than three inches in diameter would be excessive in a long flight, and that in a mean to enabling pilots to fly more hours per day with confident the dials should be enlarged.

Standards AND Lighting

He also expressed himself as strongly in favor at present of a cross-shaped mounting arrangement for instruments with all the pointers standing horizontally on the group of instruments on the horizontal line, vertically on those on the vertical line, under normal flight conditions, and all the instruments in a row tending to divide in the same direction on deviation from normal flight. That arrangement has already been employed on the instrument boards of certain machines planned for especially long flights and long periods of flying.

Turning to wheels and tires, the meeting heard and approved a recommendation from Mr. R. J. Jensen of the U. S. Rubber Company, chairman of the subcommittee on wheels and tire parts, that there should be a minor modification in the ten-inch and twelve-inch tires to secure easier mounting of tires. Mr. John R. Cuddey of the Bendix Brake Company was asked to comment on the possibilities of international standardization of tires and wheels. Mr. Cuddey who had recently returned from Europe, indicated that the problem of international agreement was one of great difficulty, although the advantages resulting, especially to operators in countries where both American and European airplanes are in quantity, would be so great that no possible of reaching a basis of agreement should be overlooked. He pointed out that the British tire industry was largely organized on the basis of millimeter sizes for its product, but that some of the tires of metric dimensions were so nearly identical with tires of such dimensions used in this country that interchangeability might be possible. He ended the discussion of the meeting further to the fact that our British tire manufacturers have 72 different sizes and forms of airplane wheels and 29 different sizes of tires in stock, and that the drop-center rim and straight-side tire have made no such headway in Great Britain as in this country, British airplanes will be designed for either size in many instances.

Aluminum Hardware

ANOTHER ADDRESS from Mr. Goss of the Consolidated Aircraft Company pointed against the present standard for tie-bolts in permitting excessive bending pressure for carbon steel, with frequent elongation of the bolts in the legs as a result. If C. B. Boswell of the Keystone company agreed that the present ones were too light, especially when aluminum alloy legs were used, J. M. Johnson of Wright Field reviewed their specifications and recommended that the present ones be modified in many ways in order to assist the use of aluminum alloys in fittings, but said that he had encountered



Some of the equipment prepared for use in the aircraft lighting tests. The light on the left is the special "bottle" size.

terrible diagnosis of the holes in carbon steel fittings only when the number was one subject to reversing loads, never when it was a wire in pure tension. He suggested that there should be further study of the loading conditions in those members, especially when a standard terminal was used in attaching a wire to a fitting.

Mr. Goss had further pointed against the present process of making aircraft bolts only in lengths spaced one-eighth inch apart. He found it often to require jacking out with wrenches to an undesirable extent in order that the bolt thread might not be placed in bearing and that the nuts might secure a sufficient grip on the bolt. Mr. Boswell again concurred, and J. F. Harlocker, Jr., who represented the Naval Aircraft Factory at the meeting, declared himself personally agreeable to reducing the spacing of lengths to one-eighth inch intervals, doubling the number of standard hole lengths in each diameter. Mr. Harlocker further spoke of the importance of standardizing engine hardware as well as that of the airplane itself. The Army and Navy, he said, are working upon the subject now, and are eager for the cooperation of the Society of Automotive Engineers and of the whole commercial engine industry.

Mr. F. W. Caldwell, chief engineer of the Standard Steel Pipe Company, brought the discussion to a close by urging that a standard should be adopted for a taper propeller shaft not smaller than the standard DX-5 shaft which stands at the foot of the present list of S. A. E. standards at that field. He had found that many low-powered engines were being designed with crankshafts of too small a diameter to permit the use of the DX-5 or S. A. E. No. 3 shaft end and that a larger number of manufacturers were in present proceeding each with its own design, with danger of harmful conflict and increase of expense for special propeller shafts.

maximum and minimum speeds were determined by the formulas

$$V_{max} = 1.5 \sqrt{P/S} \sqrt{W}$$

$$V_{min} = 16.5 \sqrt{\frac{W}{S}}$$

which although far from infallible, have been found to give a satisfactory approximation of the average results obtained with good modern designs. W , S , and P respectively, are the gross weight, wing area in square feet, and horsepower. It was further assumed that two-thirds of the load remaining after the structure of the power plant and the fuel are taken out can be considered as payload, the other third going for the crew, navigating instruments and radio, and the furnishings of the cabin. Calculations were then made for the payload per horsepower that can be carried, and the cruising that can be expected in planes of good present-day design for three different minimum speeds. The results are tabulated below, and are also plotted in Fig. 1, where cruising times have been drawn in dotted. The upper series is for a ceiling of at least 10,000 ft. as the minimum of safety for commercial airplanes means a machine without enough reserve of power to reach that height would be unable to fly at even a couple of thousand feet above sea level with one out of three engines dead. If it is to be necessary to operate over mountainous country, the required ceiling should be increased to at least 7,000 ft. above the highest altitudes that have to be crossed.

The suggestion has frequently been made, although less often now than a few years ago, that a commercial airplane ought to carry at least five pounds of pay load per horsepower. Fig. 1 shows that, so long as the performance figures used in the calculations remain valid, load

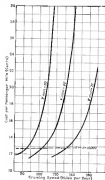


Fig. 1

capacity can only be combined with a ceiling and reserve of power adequate for safety by designing for a minimum speed of about 80 m.p.h. and a cruising speed of about 100.

Turning to the economic aspect, the paper included the formulas

$$\text{Engine} = 14 P$$

$$\text{Airplane} = 8 P + 12 W$$

for the cost of the engine, including its accessories, and the cost of the airplane, including the accessories but not the power plant, respectively. P is the formula for the total horsepower. W is the gross weight carried. These figures were intended to check reasonably with the average of present practice. In leaving out those elements which would be independent of the speed of the airplane, we entirely neglected. There remained for consideration fuel and oil consumption, depreciation, insurance and interest as the original investment. Insurance was reduced to 36 per cent, depreciation on the basis of a life of four years of continuous flying for the airplane and two and a half for the engine, and interest at eight per cent. Maintenance was taken at a figure corresponding approximately to \$1.00 per hour of flight on a 150 hp. engine, and at just about 1/2 of the value of the airplane structure and fuel, assuming 800 hours of flying per year. Fuel and oil were figured at average present prices. It was then easy to figure the variable costs of operation in terms of cents per passenger mile. In Table II, reproduced from the paper that has been cited with the assumption of an average passenger load of two-thirds capacity, and with an allowance for overhead and other fixed expenses not included in the calculations of true costs per passenger mile, we bring the total cost to about the level of present fares. The

same data are plotted in Fig. 2, where the line corresponding to a 10,000 ft. ceiling is again included.

The paper called attention to the rather remarkable fact that for a given ceiling and so for a given proportion of reserve of power a maximum operating cost per passenger mile is exactly the same for all minimum speeds and for all cruising speeds, provided the proper relation between minimum speed and cruising speed is maintained. The latter must be about 1.25 times the former for the most economical operation compatible with a safe reserve for emergency against loss of power or the necessity of maneuvers.

The writer's next step was to figure the fare for a 400-mile trip under various conditions, and plot it against time. The results reproduced in the paper is shown in Fig. 3. From that in turn it was possible to calculate the cost per mile saved by increasing speed, and the results appear in Fig. 4. Thus if a plane with a minimum speed of 80 m.p.h. and a cruising speed of 100 is said to be unacceptably slow, and if it is desired that the speed be increased, it must be recognized that the first saving in time will have to be purchased at the rate of about \$16 per hour, a rate which will increase steadily as the speed increases.

By increasing the minimum speed or landing speed, however, cruising speed can be increased at a constant cost. It was considered that if higher cruising speeds, such as are often proposed, are to be realized in practice it must be done by:

- (1) Increasing the aerodynamic efficiency of the design so that the performance formulas used herein will no longer apply.
- (2) Increasing the structural efficiency, either of air frame or engine, to reduce weights below present practical limits.
- (3) Reducing such items of expense as insurance and depreciation.
- (4) Using a variable-lift mechanism of some type for the wings, or
- (5) Raising the landing speed.

The fifth, third, first and second of these possibilities were indicated to be of that order of promise for the near future. The limitation upon the increase in landing speed is one of safety, especially important when landing has been reduced to a possibility. "In ordinary case," the paper announced, "those seeking to operate at a cruising speed of more than 1.8 times the minimum flight speed appear economically unviable at present." To increase cruising speed beyond the present level without such restrictions in design or materials (without such restrictions in design or materials increases in efficiency) we must therefore increase landing speed, and the possibility of taking that desirable step depends upon the use of engines, engine combinations, and maintenance methods that will provide a velocity actual sufficient against emergency forced landings and upon the provision of airports large enough and smooth enough to permit of landing and taking-off at very high speeds without undue difficulty or danger. These are the keys of the problem.

The discussion was confined to remarks by Capt. E. S. Land, chairman of the committee, who took exception to the advocacy of higher landing speeds, and urged that it was desirable either to work for lowering them or the interest of safety. The writer replied that he was as heartily second with that view, as it affected airplanes

likely to be flown by pilots of limited skill, but that he considered that the special all possible landing speed should be wider than at present. Instead of making to keep landing speeds around the margin of 30-60 m.p.h. in every case, they might well be set at 40 m.p.h., with a wing loading of 6 lb. per sq. ft. or less, for machines for the private owner and allowed to rise to 70 or even more for multi-engine transport machines operating between metropolitan airports.

A preliminary feature of the dinner, before the technical meeting began, was the award of the Miley Memorial Medal to Samuel D. Hixson. The Miley Medal was founded by the Society of Automotive Engineers in memory of Charles Matthews Miley, former

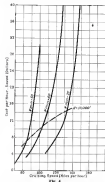


Fig. 2

president of the Society, chairman in charge of the construction and test of the Langley flying machine of thirty years ago, and especially noted for his design of the engine in the Langley plane, which produced approximately 30 hp on a weight of 120 lb. at a time when most available engine designs would not have been prepared to reach one-fifth of that power weight ratio. It is considered usually for the best paper on an aeronautical power plant subject presented before the Society during the year, and was to Mr. Hixson, formerly of the Royal Aircraft Establishment and now of Wright Field, for his paper on non-rigid air-cooled engines, delivered at the Chicago meeting last December. In making the address of presentation at the dinner, Charles L. Lawrence, one of the three judges who selected the winning paper, paid tribute to C. M. Miley's contributions to the art of aircraft engine design, and spoke with warm appreciation of the value of the work on air-cooled cylinders done by Mr. Hixson at Wright Field in number of years since the pioneer days of American radial engine development.

FOREIGN ACTIVITIES

Air Recommendations
Made at Amsterdam

AMSTERDAM (REUTERS)—A series of important recommendations relative to international aviation matters were adopted at the meeting here during July of the International Chamber of Commerce Institute on the Transatlantic passage since: (1) That governments with an ample delay as possible, convene a new international conference in 1941, proposed by the International Technical Committee of Experts on Air Laws relative to the problem of establishing uniform regulations on the liability of the air carrier. (2) That governments in drafting and interpreting international conventions on air navigation should refrain from placing any barriers in the way of the international growth of commercial aviation as air traffic. It is further recommended that national legislatures be advised to a minimum.

(3) That free airports be built in the principal international trade centers and that accommodations be made there for the loading, unloading and unloading of goods, and that remaining according and equipping of places used in regular transport be made free of all customs duties. (4) That every encouragement be given to the development of international air mail plans, and that the documents issued by the Universal Postal Convention in London allowing modification of the Universal Postal Convention relating to air mail be applied uniformly wherever international aviation. (5) That the various governments, particularly in Europe, and interested parties within those areas, foster direct negotiations between air transport companies and national agencies for the development of air mail projects. Urgent attention is called also to the matter of coordinated air and sea routes which is of growing importance.

Mexico Bans Air Routes

MEXICO CITY (REUTERS)—The Mexican Post Office Department is helping along publicity for the newly established air mail and passenger service between Mexico City and Tijuana, Lower California by posting posters in post offices throughout the Republic announcing that the service requires mail to reach Los Angeles, Calif., two days after leaving the Mexican capital. This line, which is operated out of here three times a week, places express reports Mexican cities on an air route. The service is conducted by the Latin American Air Transport Company, Mexico and other business houses in the larger cities of Mexico are also looking a hand toward stimulating interest in the country's air mail services.

Start South African Mail Line

CAPE TOWN (SOUTH AFRICA)—A new air mail service is being operated between this city, Port Elizabeth, Johannesburg and Durban in South Africa, in collaboration with the arrival and departure of mail aeroplanes on the service to England. Almost two days time is saved in transmission of mail to Durban. The South African Government has been encouraging air mail and air transport services in recent months and such along this line is expected to be realized through the operation of a company similar to the National Flying Service, of England.

Metal Monoplane
Is Built in Australia

MELBOURNE (AUSTRALIA)—With a design specially adapted to Australian requirements, the "Lancaster" 3-6 passenger high-wing, semi-cantilevered metal monoplane has recently been put on the market by Luton Aircraft Supply Company, Ltd. A chassis of 5 ft. 6 in. under the wings, and an undercarriage with 13 ft. track together make possible side loadings in the scrubs or grass-covered land which forms such a large proportion of the country.

The new plane is so designed that it can be taken apart and re-assembled very easily. Front end doors, and rear end of the fuselage are built in distinct, easily detachable sections, and the wings are also easily detachable. Special provision for passenger comfort have also been made, such as adjustable reclining, double thickness vinyl-covered cabin walls, and semi-pneumatic upholstery.

Fitted with a "Puma" engine the plane has a cruising speed of 89.80 m.p.h., and a range of about 600 mi. It has independent wheel brakes, and is otherwise very completely equipped, with many more optional accessories to the purchaser's preference.

Build Large English Flying Boat

LONDON (REUTERS)—Following the recent test flights of the Avro Lincoln, a four-engine flying boat, the importance of improved construction of a large flying boat is being recognized in Germany. It has been reported here that the Blackburn Aeroplane & Motor Company, Ltd., of Yorkshire is building under contract a four-engine flying boat, which will seat 30 persons and have a range of 1,900 mi. without refueling. Construction is under way in the Blackburn Works (located in the Avro 34 house of Avro). The new aircraft will be powered by three Rolls-Royce engines.

Foreign News Briefs

The twenty-second meeting of the International Air Traffic Association, was held at the Hague Aug. 20-22 and was attended by delegates from 23 European air navigation companies.

An international show for light planes will be held at Paris September 25-30 under the auspices of the French Aero Club.

Plans are under consideration for a central airport in the growing suburbs at Stuttgart, accommodations to include a meeting point for arrivals of the projected Berlin-Tokyo and London-Hankow routes, financial means to be provided by Germany, France and Berlin.

30 ft. Long Berlingue established a new record for light planes in last month's flight at 1,562 mi. from Mantes, France, to Solferino, Egypt.

The Western Australian Aircraft Ltd. reports weekly traffic of 23,000 letters over its air mail routes.

Plans released claimed the world would see busy planes by crossing the 965 mi. straight course in 5 1/2 hr. near Buenos Aires, on Aug. 30.

A trans-Atlantic Robinson-Royce flying boat was recently given a three-flight test by the Luft Hansa over the English Channel and the North Sea.

The Southern Cross is being considered at the Amsterdam Folkestone route for the proposed flight from Amsterdam to America.

The British Air Service, Company has just completed plans for surveying the operations of the 50 ft. at a total cost of \$100,000.

The position of the members of the Irish Free State Army Air Force and the Commercial Air Service has usually been recently proposed.

Various newspapers in Zurich the Dutch broke water, are racing funds for just acquisitions at Brussels of an airport. A 100 ft. 150 acres has already been secured.

Flight tests have been arranged at the Berlin and Amsterdam which will include provision for construction in Britain.

Eighteen large aircraft carriers, the German, is nearing completion and will be assigned to the Mediterranean. They will be the second line has been ordered by the Atlantic fleet two in the North Atlantic and one in China waters.

The Junkers "Stinson" type monoplane is being sent to Germany for further testing.

The Imperial Airways Ltd. flights over London are said to be proving somewhat disappointing.

THE
AERO
LIGHT-
HOUSES

The aero lighthouses with their shafts of light piercing the darkness at regular intervals are an indispensable aid to night flying. The beacons used on these towers must be so constructed as to be unflashing in their operation under any and all weather conditions.

The Crouse-Hinds Company manufactures a very rugged and dependable revolving beacon which is the government standard for use on aerolights.

Type DCB34 revolving beacon shown above is of cast aluminum sheen non-corroding alloy. A lamp changer can be provided for automatically throwing into focus a spare lamp upon failure of the first one. The beacon can also be furnished with search light which enable an aviator to locate the beacon after he has passed over the main beam.

Everything in Airport and Airway Lighting Equipment

CROUSE-HINDS

CONDUITS - GROUNDWIRTS - PLUGS and RECEPTACLES - FLOODLIGHTS - TRAFFIC SIGNALS
AIRPORT and AIRWAY LIGHTING EQUIPMENT - PANELBOARDS and CABINETS - SWITCHES

CROUSE

"BIRD"



Why they suit the DEALER!



Safety in Training—Speed in Transportation DEPENDABILITY at all Times

THE MORE SELLING of a plane is NOT ENOUGH. Many times the sale depends upon instruction; often a sale is made to a newly qualified pilot. The dealer's problem is to sell the plane which will afford the quickest instruction and at the same time assure the new pilot of the highest degree of personal safety.

And again, to sell for transportation means the selling of speedy miles, ease of handling, and low cost of operation.

BIRD dealers are satisfied that these planes embrace the features of ALL THAT CAN BE DESIRED IN THE FINEST OF AIRCRAFT.

"MORE PERFORMANCE PER HORSEPOWER"



PLANES



Why they suit the FLYER!

1. Choice monodydium tubing throughout.
2. Inherent stability perfected to a degree which allows fool-proof student training.
3. Dual control with front seat quickly detachable for passenger carrying.
4. Perfection in aerodynamic design.
5. Wing design perfected aerodynamically to a degree which permits performance comparable to dated wings.
6. Landing gear of split axle type with combination steel and rubber shock absorbers.
7. Metal turtle deck from front to rear allowing internal inspection.

WHEN a BIRD pilot grasps her the throttle he knows that he can **OUTSPEED** any other class of the same horsepower.

When he hovers over a small field he knows that he can **LAND SAFELY**—and **GET AWAY** with EASE.

When he takes a passenger—whether for hire or as a guest—he knows that his passenger is afforded the **HIGHEST DEGREE OF SAFETY**.

When he lands in a strange field he need not apologize for his plane, because he knows that he has just brought in the **GREATEST PERFORMER PER HORSEPOWER**.

Confidence and pride are manifested in the satisfaction with which the pilot regards his BIRD.

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The remarkable industrial growth of Los Angeles County may be attributed to:

Natural year-round climatic advantages

Tremendous population growth

High per capita buying power

Abundance of trained labor

Low building costs

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Largest concentrated market in Pacific Coast

Splendid transportation facilities

Economical access to Pacific Coast and export markets

Air-Minded
→ **LOS ANGELES COUNTY**

Complete detailed survey and information supplied upon request by INDUSTRIAL DEPT., LOS ANGELES CHAMBER OF COMMERCE

Flying conditions are as nearly ideal as the year-round in Los Angeles County as anywhere in America.

U. S. Weather Bureau reports over the last 50 years show an average wind velocity of 3 miles per hour . . . 155 days per year with sunshine . . . 154 days per year when the temperature is neither above 80 nor below 40.

17 manufacturers of airplanes and 9 manufacturers of airplane motors have already located here.

There are 25 or more aviation schools, 1200 aviation students, and more than 3,000 pilots in Southern California.

6 passenger transport companies operate 51 regularly scheduled lines out of Los Angeles, which is the terminal for 4 coast-to-2 coast lines.

These are facts which the manufacturer of airplanes, motors or parts can turn to his profit . . . for these conditions have a tremendous effect on production costs and successful operation . . . and they are not paralleled elsewhere in America.

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Perfected IRELAND AMPHIBION . . . most modern of all Air-Yachts

CONSIDER the amazingly comfortable five-place "Neposun" . . . an all-weather Air-Yacht . . . it carries you quickly and safely at any time, of year to any destination on land or water.

The large, commodious cabin is complete in every detail of appointment . . . and it's so quiet that you can talk in conversational tones. The adjustable, unbreakable glass windows give perfect light and vision . . . or to those who enjoy the thrill of flying in the open air the Neptune comes with open cabin.

In all-around performance the perfected "Neposun" is the highest achievement of Amphibious construction. Outstanding exceptional maneuverability — strength — flexibility — power, speed, and ease of control . . . its economy of

operation and low upkeep cost are strong factors in its rapidly growing popularity.

Remember, too, in a Neptune or any



The Neptune lands easily and rises quickly from land or water

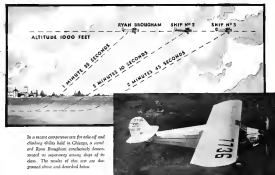
one of Curtiss' wide range of planes, you will find country-wide service and accommodations. For Curtiss has monopolized the service and spareparts of the country with 80 branches and 53 dealers.

Write or call at "Aviation Branch quarters," Curtiss Flying Service, Dept. 6, 17 West 25th Street, New York City, and we will get you immediately the nearest Curtiss Branch. There you can arrange to fly the Neptune . . . subject it to every test you want, and learn first-hand its true distinction.

Sales agents for Curtiss Robertson Airplane Mfg. Co., General Aircraft Co., Curtiss Aeroplane and Motor Co., Looperford, Indian Aircraft, International, Catalina-Air, Inc., Moth Aircraft Corporation.

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FOR TAKE-OFF AND CLIMB —BEST INSURANCE IS A RYAN



In a recent comparison for take-off and climbing ability held in Chicago, a vessel of Ryan's design, conclusively demonstrated its superiority among ships of its class. The results of this test are diagrammed above and described below.

CARRYING six people and 100 gallons of gasoline, a standard Ryan 85 Brougham took off from a Chicago Airport and climbed to an altitude of 1,000 feet in 1 minute and 30 seconds.

Step No. 1, in competition with the Ryan, carried six people and 97 gallons of gasoline and required 2 minutes and 10 seconds to take off and gain a height of 1,000 feet.

Step No. 3, with load of six people and just 85 gallons of gasoline, took off and reached 1,000 feet altitude only after 2 minutes and 45 seconds.

This marked victory of the Ryan Brougham—not in a "trial" demonstration, but in a competitive test with an imperial buyer as judge—furnishes further proof that the Ryan will outperform all other ships of equal load and power.

Powered by the new Wright 300 H.P. 36 White-wind motor—with installation properly related to the ship's design—today's Ryan will take off fully loaded in 175 feet, and in 8 seconds time, will climb at the rate of 1,000 feet a minute, and will land in a 100 foot circle. It provides top speed of 140 miles per hour and cruising speed of 120 miles per hour. Cruising radius is 700 miles.

Write for illustrated catalog describing details of Ryan design and construction. Learn why Ryan's performance and safety records are unrivaled by ships of its class.

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SISTER SHIP OF THE "SPIRIT OF ST. LOUIS"

Department of Commerce Approval Type Certificate No. 124,
issuing load, gear and passenger

FULL 100 HORSEPOWER

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AIRPLANE AND
MOTOR CORP.
GLENDALE, CALIF.**



This KINNER Message is addressed to Manufacturers of 4 and 6 Place Planes



THREE Kinner Five Cylinder 500 Horsepower Motor has proven its dependable performance. Now, thirty-five manufacturers of two and three place airplanes are powering their ships with Kinner Motors!

The same degree of dependable performance, that has made Kinner America's standard light plane power plant, is offered

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Three Kinner motors, developing a factory-guaranteed total of more than 300 horsepower affords a greater safety factor...in emergency...with sustained flight assured on two motors, while under the most remote possibility, one motor materially lengthens the gliding range!

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CARROLL, CALIFORNIA

AVIATION
September 7, 1930

31

BLAW - KNOX patented STEEL HANGARS are the result of specialized engineering research to determine the most economical type of interior construction to fit all requirements of support or loading field.

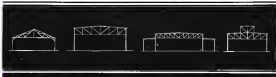
Over twenty years' experience in the construction of Standard Steel Buildings has resulted in Hangars which are—

1. Made from stock sizes, easily erected, enlarged, changed in shape or moved at occasional intervals.
2. Fireproof.
3. Weather-tight and easily heated.
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with the new
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The Russell "Lobe" Parachute fits right into the seat—and the new detachable BACK-PAD holds the harness in position while forming an additional cushion for greater comfort.

The harness and sections the parachute become when you are seated in the cockpit on the same manner as you may the whole unit. It is not only simple and practical — it is the solution of the many other problems you will find in the Russell "Lobe" Parachute.

Formerly light, air transport companies, airlines, schools, and all kinds of men and women recognize the world over have adopted the Russell "Lobe" Parachute as standard flight equipment.

In an emergency you simply pull the release ring — the Russell "Lobe" Parachute descends the seat — it has a record of 100% efficiency in use!

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Ask about the new Russell "Lobe" Parachute 500 Parachute, selling at \$275. Other Models, \$250 to \$350.


Amelia Earhart and Frank on Russell.



Ribs are punched into cup shape of rib; fabric is applied to fuselage, under tail over each hole and self-tapping screws inserted in. Then round fabric patches are slipped over the screwheads.


A new method of attaching fabric

Consolidated Aircraft develops this secure, quick and easy way of fastening fabric to the metal ribs of the Fleet biplane . . .



HARDENED SELF-TAPPING SHEET METAL SCREWS

For making sheet metal assemblies. Turn the Hardened Self-tapping Sheet Metal Screws into a parent or drilled hole. As the Screw is driven, it cuts its own thread in the metal like a tap, drawing sections securely together.



HARDENED METALLIC DRIVE SCREWS

To make permanent fastenings in wood, brass and aluminum castings, steel, bakelite etc. Insert the Hardened Metallic Drive Screw into a drilled or formed hole. The Screw taps a thread in the material, and makes a fastening that stays secure even under constant vibration and severe service.

The design of the FLEET biplane made it impractical to attach fabric to the metal ribs of the wing and tail surfaces by the usual cord-lacing method. A new way had to be found. Consolidated Aircraft engineers used Hardened Self-tapping Sheet Metal Screws, which proved to be an ideal means of effecting the desired results. Not only did these Screws assure secure fastenings but they provided an easy, quick and economical method.

This is but one of the many assemblies for which Hardened Self-tapping Sheet Metal Screws and Hardened Metallic Drive Screws have been adopted by prominent aircraft manufacturers.

These unique Screws eliminate costly tapping operations . . . speed up production . . . cut the cost of making fastenings. And, in some instances, they have made possible simplification and improvements in aircraft design.

We suggest a test of these Screws on your own work. See what they will do . . . what they can save you. A brief description of assemblies to which you think these Screws adaptable brings suitable samples, free.

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Equipped with the highest priced bearing in the world

YOU MAY BUY A
BEARING AS A
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TRY AND GET A
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Antarctic and Arctic research
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Savings on Bearings are Apt to Prove Costly in the Air—MacClatchie Uses **SKF**

WITH aviation making rapid strides
towards becoming one of our
major industries, safety and reliability
assume greater importance daily. And
the building of confidence is not based
on the saving of pennies in equipment.
That is why, practically the entire aviation
industry uses **SKF** Bearings.

The MacClatchie Mfg. Co. is one of
numerous manufacturers that subscribe
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Anti-Friction Bearings in vital locations
on their Panther motors make certain
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Equipped Aviation Schools



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training is completely complete
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A flying school can be no better than its equipment and personnel. No school in the country offers finer or more modern equipment than the Yellow Cab Airways SCHOOL OF AVIATION. Our training fleet comprises two Fairchild Wasp Seven Place Cabin Monoplanes, a Five Place Ryan Broughane, a Four Place Monocouche, a Monocoupe, a Curtiss-Robin, two Travel-Airs, a Waco, and two First Training Planes, with three more on order. Every instructor holds a government license.

Complete Flying and Ground Courses

Our courses will qualify you for a Private Pilot's, Limited Commercial, Transport Pilot's or General Mechanic's License. Because of our careful supervision for safety we are able to offer our students complete insurance protection throughout their training.

Write for FREE Catalogue—"Sky Riders"

\$15,000 all-steel hangar, completely equipped shops, special night-flying equipment. Located on two Transcontinental air lines. Large, level field. Board and room as low as \$8 to \$10 a week right in the school. Every student flies every day, weather permitting. Our rates are far below all other schools offering such high class, complete training. Write at once for free catalogue, references, rates, etc.

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3. Colonial is identified with the earliest operation of the Air Mail. Contract Air Mail Routes 1 and 20, and Foreign Air Mail Route 1 are all operated by Colonial companies. Passenger Transport Lines between New York and Boston, New York and Montreal and Buffalo and Toronto are a part of the Colonial System. Colonial Flying Schools are maintained in New York State and New England.

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